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The influence of low-temperature surface induction on evacuation, pump-out hole sealing and thermal performance of composite edge-sealed vacuum insulated glazing

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	ACCEPTED MANUSCRIPT	
1	The influence of low-temperature surface induction on evacuation, pump-	
2	out hole sealing and thermal performance of composite edge-sealed	
3	vacuum insulated glazing	
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#### Abstract

- Hermeticity of vacuum edge-sealing materials are one of the paramount requirements, specifically, to the
- evolution of energy-efficient smart windows and solar thermal evacuated flat plate collectors. This study
- reports the design, construction and performance of high-vacuum glazing fabrication system and vacuum
- insulated glazing (VIG). Experimental and theoretical investigations for the development of vacuum
- edgeseal made of Sn-Pb-Zn-Sb-AlTiSiCu composite in the proportion ratio of 56:39:3:1:1 by % (CS-186)
- are presented. Experimental investigations of the seven constructed VIG samples, each of size
- 300mm 300mm 4 mm, showed that increasing the hot-plate surface temperatures improved the cavity
- vacuum pressure whilst expediting the pump-out hole sealing process but also increases temperature
- induced stresses. Successful pump-out hole sealing process of VIG attained at the hot-plate set point
- temperature of 50°C and the approximate cavity pressure of 0.042 Pa was achieved. An experimentally and theoretically validated finite volume model (FVM) was utilised. The centre-of-pane and total thermal
- transmittance values are calculated to be 0.91 Wm<sup>-2</sup>K<sup>-1</sup> and 1.05 Wm<sup>-2</sup>K<sup>-1</sup>, respectively for the VIG. FVM
- results predicted that by reducing the width of vacuum edge seal and emissivity of coatings the thermal
- performance of the VIG is improved.

Keywords: vacuum; glazing; solar-thermal; performance; modelling; transmittance

#### 1 1. Introduction

2

3 Advancement in the vacuum sealing materials is one of the paramount need in leading smart windows [1] 4 and solar thermal evacuated flat plate collectors [2] at the manufacturing level due to considerable issues of 5 leakage in the vacuum edge sealing materials [3] and/or the cost of scarce semi-precious materials such as 6 indium [4, 5]. There is also a serious challenge particularly in solar energy field of balancing the security of 7 power supply and peak power demand [6]. Glazing technologies, such as double air-filled glazing [7] with 8 low-e coatings [8] and gas filled glazing with cavities filled with heavy gases (Argon, Krypton or Xenon), could achieve the thermal transmittance value (U Value) up to 1.4 Wm<sup>-2</sup>K<sup>-1</sup>, depending on the cavity 9 thickness [9, 10]. To improve the thermal performance further, without compromising the visible light 10 11 transmittance, a vacuum insulation is an option. A vacuum insulation is a space, between two glass panes, 12 of reduced mass of atmospheric-air. The rate of decrease of the density of air in a space determines the 13 level of vacuum pressure. This provides thermal insulation, because with a lower density of air the mean 14 free path between air molecules can be increased to above 1000 m [11], ultimately reduces the heat transfer path between air molecules in a space. In VIG, the space between two glass panes is evacuated to high-15 vacuum pressure (0.13 Pa to  $1.33 \cdot 10^{-4}$  Pa) in order to reduce conductive and convective heat transfer [12] 16 to negligible levels, however the heat transfer through radiation can only be minimised using low-emittance 17 18 coatings to VIG [13]. In evacuated flat plate collectors, selective anti-reflective emissivity coatings onto the glass surface are required that improves optical transmission which is different to VIG in itself. Due to the 19 20 difference between external atmospheric-air and internal vacuum pressure, spacers are required to prevent 21 the glass panes touching each other [14]. These spacers are called support pillars and typically have radii 22 from 0.1mm to 0.2 mm and height of 0.1mm to 0.2mm [15]. In VIG, even a small vacuum space gives the 23 same thermal insulation because radiative heat transfer is same at any cavity thickness [16]. A vacuum 24 edge seal around the periphery of the glass panes is required to maintain the high level of vacuum and 25 avoid the problems of gas leaks, degradation of coatings, and absorption of moisture. However, heat transfer through conduction occurs because of the contiguous heat transfer path formed by the support 26

- 27 pillar and edge sealing materials.
- 28

29 The constructional components that mainly determines the thermal performance of VIG is its vacuum edge

- 30 seal [17,18]. The vacuum edge seal of a VIG must be capable of maintaining a vacuum pressure of less
- 31 than 0.1 Pa [19], in order to suppress gaseous conduction, for the expected life of 20 years. The edge of two
- 32 glass panes was first sealed using a high power laser through a quartz window in a vacuum chamber [20]
- but the level of vacuum was not less than the required, 0.1 Pa, due to gases and vapour molecules caused
- 34 by laser sealing technique [21, 22]. A high-temperature edge sealing material, Schott solder glass type 8467
- at the sealing temperature of 450°C, was used by the group at the University of Sydney [12, 23, 24]. With
- 36 this technique, it achieved centre-of-pane thermal transmittance ( $U_{centre}$ ) value of 0.8 Wm<sup>-2</sup>K<sup>-1</sup> and
- 37 subsequently developed to the production level under the trade name of 'SPACIA' in Japan by Nippon
- 38 Sheet Glass (NSG) [25]. The problems with the high-temperature edge sealing method is that it causes
- degradation of soft low emittance coatings meaning that only hard coatings can be used [13]. Toughened
- 40 glass also cannot be used due to the loss of temper at high temperatures [26]. Low-temperature solder glass
- 41 materials were investigated to form a hermetic edge seal, but durability was a problem due to the
- 42 absorption of moisture. Polymers have problems of both gas permeability and out gassing [4, 27]. A low-
- temperature edge sealing materials, i.e. indium or indium alloys melts at about 160°C, were utilised and
- 44 developed at the University of Ulster [13, 28, 29]. This technique achieved a  $U_{centre}$  value of 0.9 Wm<sup>-2</sup>K<sup>-1</sup>

and allowed the use of low emittance soft coatings (such as silver), which reduce radiative heat transfer
between the glass panes and permits toughened glass pane for an increase of support pillar spacing that
reduces conductive heat transfer. The problems with the low-temperature based indium seal are the scarcity
and the cost; because of this, the low-temperature indium sealed vacuum glazing process has not yet been
commercialised [4,9, 30].

6

7 In this paper, a particular focus is made on the design and construction of high-vacuum glazing fabrication

- 8 system, including the modified vacuum cup, and a new method of vacuum edge seal utilised for the
- 9 successful fabrication of the VIG, made of Sn-Pb-Zn-Sb-AlTiSiCu composite in the proportion ratio of
- 10 56:39:3:1:1 by % weight respectively, developed by MBR Electronics Gmbh in the trade name of CS-186.
- 11 A steel reinforced epoxy applied to support the vacuum edge seal, as illustrated in Fig. 1. One of the
- 12 significant contribution in this paper is reporting the experimental investigations of seven VIG samples for
- evaluating the influences of hot-plate surface temperatures induction on evacuation and pump-out hole
  vacuum sealing of the VIG in order to achieve the relatively acceptable setup when the evacuation and
- 15 pump-out hole sealing processes are performed. An experimentally and theoretically validated finite
- 16 volume model (FVM) of Fang et al.(2005) [31]; Fang et al.(2006) [15] and Fang et al. (2009) [22] was
- 17 utilised for the thermal performance analyses of VIG, size of 300mm 300mm 4mm rebated by 10 mm in a
- 10 and forme and 10 mm mains in the state of subminity subminity and the state of subminity and the st
- 18 solid wood frame and 10 mm main edge seal and the results are discussed.



Fig. 1. A schematic diagram of novel edge sealed VIG showing the main vacuum edge seal 10 mm wide,
made of Sn-Pb-Zn-Sb-AlTiSiCu composite in the proportion ratio of 56:39:3:1:1 by wt% respectively )
(CS-186), and a support edge seal 4 mm wide, made of steel reinforced epoxy.

40

#### 41 2. Design and construction of a high-vacuum glazing fabrication system

42

43 A lab scale vacuum glazing fabrication system was designed and constructed to fabricate VIG. The vacuum

44 glazing production system design, as shown in Fig. 2, consists of the vacuum pump, it is connected in

series with the vacuum cup. For the measurement of pressure, a pressure gauge is connected in parallel

- 46 with the vacuum pump. The angle valve with Swagelok adapter is included allowing the system to be
- 47 purged with nitrogen (inert gas); this is connected in series with a square cross-section tube. An angle valve

- 1 is connected in series between the vacuum pump and vacuum cup. The dimensions of the components used 2 in this design are presented in Table 1. A dry type turbo-molecular with backing pump with an achievable pressure of  $5 \cdot 10^{-6}$  Pa was chosen. This is because the vacuum pump should be of an oil free/dry type as the 3 4 contamination in the oil type with oil molecules could occur on the surfaces of tubes, valves, hose and/or 5 vacuum cup preventing an achievement of effective vacuum level. A turbo-molecular vacuum pump has a 6 pumping speed of 61 litres/sec. With proper venting, the turbo mechanism stops in less than a minute. This 7 means that vacuum cup venting is accomplished without the need for a valve to separate the pump and 8 vacuum cup. The EXT75DX T-Station selected for this vacuum system. It consists of a turbo molecular 9 pump and a diaphragm-backing pump XDD1. The ATV (Atmosphere to Vacuum) transducer type 979, was 10 connected to a PDR 900 digital pressure measurement readout, used in the present study for the 11 measurement of vacuum pressure in the designed vacuum system. This pressure gauge is located at the 12 closest possible location to the vacuum cup to measure the approximate pressure in the cavity of the VIG, 13 as shown in Fig. 2. The ATV transducer enables measurement of a wide pressure range from ultrahigh vacuum (1.33·10<sup>-8</sup> Pa) to atmospheric pressure (101.33·10<sup>3</sup> Pa). It consists of MEMS (Micro-Electro-14 15 Mechanical System) based MicroPirani gauge and a miniaturised hot cathode ionisation gauge in a single transducer unit [32]. The MEMS based MicroPirani gauge measures pressure from  $1.33 \cdot 10^{-2}$  Pa to 16
- transducer unit [52]. The MEMIS based MicroPiram gauge measures pressure from 1.55-10 Pa to
- atmospheric pressure. The hot cathode ionisation gauge measures pressure from 1.33 · 10<sup>-3</sup> Pa down to
   1.33 · 10<sup>-8</sup> Pa. A good discussion and literature review of the fundamental theory of Pirani and hot cathode
- 18 1.33 · 10<sup>-8</sup> Pa. A good discussion and literature review of the fundamental theory of Pirani and hot cathode
   ionisation gauges can be found in text books by Dennis and Heppell (1968) [33] and Guthrie (1963) [34].
- 19 Ionisation gauges can be found in text books by Dennis and Heppen (1908) [55] and Outline (1905) [54].
- 20 The PDR900 digital controller provides readout of the pressure measurements. It interfaced to a computer
- 21 for real time data logging of the evacuation pressure of the vacuum system.
- 2223 Table 1

Components	D(cm)	L(cm)	$V(cm^3)$
Angle valve with	D <sub>3</sub> =4.2	L <sub>3</sub> =11	152.4
Swagelok adapter	$D_4 = 3.8$	$L_4 = 4.7$	53.3
	D <sub>5</sub> =3.8	L <sub>5</sub> =4.7	53.3
Square cross-	D <sub>1</sub> =3.8	$L_1 = 4.1$	46.5
section tube	D <sub>2</sub> =3.8	$L_2 = 4.1$	46.5
	D <sub>6</sub> =3.8	$L_6 = 4.1$	46.5
	D <sub>7</sub> =3.8	$L_7 = 4.1$	46.5
Angle valve	D <sub>8</sub> =4.2	$L_8 = 11$	152.4
	D <sub>9</sub> =3.8	L <sub>9</sub> =4.7	53.3
	D <sub>10</sub> =3.8	$L_{10}=4.7$	53.3
hose/pipe	D <sub>11</sub> =3.8	L <sub>11</sub> =69	782.54
Vacuum cup	D <sub>12</sub> =3.5	$L_{12}=2$	1.1
	D <sub>13</sub> =10	L <sub>13</sub> =15	23.56
	D <sub>14</sub> =3.5	$L_{14}=2$	1.1
	D <sub>15</sub> =3.5	$L_{15}=2$	1.1

24 Dimensions of the components used in the vacuum system design.



Fig. 2. A schematic diagram of the designed vacuum system showing the dimensions and connections ofthe tubes, angle valves, the vacuum cup and the vacuum pump.

5 To obtain a vacuum in a volume of the system the density of gas must be reduced and is directly

6 proportional to the gas pressure; in practice, the gas pressure measures the level of vacuum [35]. The rate at

7 which the gas molecules are evacuated from the vacuum vessel, i.e. mass flow, determines the pressure

8 drop. The mass flow rate, M, can be expressed (in atomic mass unit of gas) by keeping the mass of gas, m,

9 and temperature, *T*, in the vessel constant as Eq. (1), 10

$$\begin{array}{l}
11 \quad \frac{dM}{dt} = \frac{m}{\zeta T}Q \\
12
\end{array}$$
(1)

13 Where  $\zeta$  is the Boltzmann constant i.e.  $1.38 \cdot 10^{-23}$  Pa m<sup>2</sup> K<sup>-1</sup> and *Q* is the gas flow rate in Pa litres/sec. This 14 can be expressed by knowing the pressure, *P*, and volume, *V*, of the gas as Eq. (2),

15  
16 
$$Q = \frac{d(PV)}{dt}$$
 (2)  
17

18 The gas flow, Q, through a vacuum vessel or hose occurs due to the difference of pressure depending on 19 the inside diameter of the tubes. The average distance any air molecule travels before colliding with 20 another molecule is its mean free path  $\lambda$  in m [34, 36]. The collisions between molecules can be calculated 21 using Eq. (3).

$$\lambda = \frac{\zeta T}{\sqrt{2}\pi P D_m^2} \tag{3}$$

25 Where, *T* is the absolute temperature of the air in K (the tubes are under atmospheric air with an ambient 26 temperature of 294.15K) and *P* is the air pressure in Pa.  $D_m$  is the gas kinematic diameter of the air 27 molecule i.e.  $4 \cdot 10^{-10}$ m, which is based on the assumption that the air molecules are smooth, rigid and elastic

**28** spheres [37].

22 23 24

1

1 The turbo-molecular pump evacuated the air molecules continuously from the tubes, components, vacuum 2 cup and cavity of the VIG. The rate at which the volumetric flow of gases evacuated from the system is the 3 pumping speed in litres/sec, in this type of turbo-molecular pump the ultimate pumping speed is given to be 4 61 litres/sec. One of the considerations was taken into account when designing the vacuum system was to 5 reduce the connections (tubes and pipe length) between the turbo-molecular pump and the vacuum cup so

- 6 as to keep the pumping speed losses to a minimum level.
- 7

8 Upon initiating a pump down the flow of air molecules, having air pressure of 101.325 kPa, was often

9 turbulent, called viscous flow regime. In which the mean free path between molecules was calculated to be

10  $56.35 \cdot 10^{-9}$  m from the Eq. (3). As the air pressure decreases the mean free path increases, having a fewer 11 air molecules in a space to make collisions with each other and the mean free path is considered to be

12 roughly equivalent to the diameter of the tube, called a laminar (transition) flow regime. In a best-case

13 scenario, when the achievable vacuum pressure is  $5 \cdot 10^{-6}$  Pa then the mean free path between molecules is

- 14 calculated to be 1142 m, called a molecular flow regime.
- 15

16 The rate of evacuation, i.e. gas flow rate, is proportional to the rate of mass of air change. In addition to 17 that, the layers of adsorbed gaseous molecules as a thin film on the internal surfaces within the tubes and

18 vacuum glazing require evacuation of six hours to achieve a good level of high vacuum pressure.

19 Increasing the temperature from 100°C could help in desorbing the layers of gaseous molecules but this

20 may cause glass bending. This increases internal compressive and external tensile stresses in the glass

panes and increases the risk of cracking of the edge seal. With a constant temperature, up to 60°C, and volume of the vacuum system the flow rate into the turbo-molecular pump  $(Q_i)$  from the vacuum system

can be written as Eq. (4),

24 25  $Q_i = S_o P_v$ 

(4)

The flow rate into the turbo-molecular pump ( $Q_i$ ) can be calculated to be  $3.05 \cdot 10^{-4}$  Pa litres/sec from the  $S_o$ ultimate pumping speed i.e. 61 litres/sec and the  $P_v$  ultimate pump pressure i.e.  $5 \cdot 10^{-6}$  Pa.

30 A high-vacuum glazing fabrication system constructed is shown in Fig. 3. It is based on the design

presented in Fig. 2. The vacuum system was experimentally tested and the minimum achievable vacuum
 pressure was recorded to be 4.35 · 10<sup>-5</sup> Pa. This deviates by 7.7% with the ultimate vacuum pressure of the
 turbo molecular pump due to tube air-flow conductances.

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Pinholes with air trapped inside

Fig. 6. (a) FIB-SEM of 20mm 20mm 1mm slide cover slip sample with Sn<sub>56</sub>Pb<sub>39</sub>Zn<sub>3</sub>Sb<sub>1</sub>- AlTiSiCu<sub>1</sub> wt%
 composite( also called CS-186 composite) ultrasonically soldered on the surface magnified at 5000x (b)
 X-ray CT cross-sectional view at the interface of the glass and CS-186 composite seal.

### 14

# 15 4. Design and construction of the VIG

17 *4.1. Four-stage design process* 

#### 18

19 The four-stage design process for the construction of vacuum edge seal is developed, as shown in Fig. 7, 20 using the high-vacuum pump-out system. Two 4 mm thick Pilkington K-glass panes of area 21 292mm 292mm (upper glass) and 300mm 300mm (lower glass) were used. The reason for using different 22 sizes of glass panes was to apply support edge seal (steel reinforced epoxy) uniformly around the periphery 23 of the VIG to support the main edge seal made of Sn<sub>56</sub>Pb<sub>39</sub>Zn<sub>3</sub>Sb<sub>1</sub>- AlTiSiCu<sub>1</sub> wt% (CS-186) composite. 24 The width of the primary edge seal was considered to be constant i.e. 10 mm and a support edge seal i.e. 25 4mm to test and repeat the experiments for the successful fabrication of VIG based on this new method. A 26 selection of 10 mm width of the edge seal was the result of experiments performed to increase the 27 mechanical stability of the main edge seal. The process achieved after rigorous experiments is detailed 28 section 4.2. 29 30





1	4.2. Construction process
2	
3	1) 4 mm thick K-glass panes were cut to the size of 292mm 292mm and 300-300mm. In the smaller pane, a
1	A mm diameter of nump out hole drilled to allow the evacuation of the cavity between the two glass panes
- -	4 min diameter of pump-out note diffied to anow the evacuation of the eavity between the two glass panes,
5	located /5 mm from the corner of the smaller glass pane.
6	
7	2) The panes of glass were cleaned with water, acetone and isopropanol followed by an initial bake-out at
8	120°C in an oven.
9	
10	3) A 10 mm wide layer of CS-186 composite was ultrasonically soldered around the periphery on the SnO <sub>2</sub>
11	coated sides of both glass panes in the arrangement, as shown in the Fig. 8. Subsequently, a square cover
12	slip of 1mm thick cutting to a size of 18.18mm was prepared for the nump-out hole sealing by soldering
12	with CS 186 composite
1/	with CS-180 composite.
15	
16	Template for uniform edge sealing
17	
18	
19	
20	
21	
22	
23	
24	
25	
27	Top glass pane
28	
29	Seal displaced 4mm from the (CS-186 composite)
30 31	glass edge area
38	
34	Fig. 8. The 10 mm wide primary seal soldered on the bottom glass pane around the periphery, displaced
35	4mm from the glass edge.
36	
37	4) Stainless steel support pillars were located on the lower glass pane using a vacuum wand as illustrated in
38	Fig. 9a. The pre-soldered upper glass pane was located on top of the support pillars.
39	
40	5) A CS-186 composite wire gasket was placed on the soldered area as illustrated in the Fig. 9b
40 //1	5) A CS-100 composite wite gasket was placed on the soldered area as musuated in the Fig. 90.
40	$()$ The second state is $\Gamma_{i}^{i}$ , $0$ and $h$ and $100^{\circ}$ C is the second state second state of the second state s
42	6) The prepared sample, shown in Fig. 9c, was heated to 186 C in the oven to join two panes of glass
43	together for up to 2 hours.
44	
45	7) A support seal, steel reinforced epoxy, was applied around the edges of the main edge seal for enhancing
46	the mechanical stability of the main edge seal, as shown in the Fig. 9d.
47	
48	
49	
50	



Fig. 9. shows (a) support pillars placing on the lower glass pane using a vacuum wand, (b) the square cross section wire made of the Sn<sub>56</sub>Pb<sub>39</sub>Zn<sub>3</sub>Sb<sub>1</sub>- AlTiSiCu<sub>1</sub> wt% composite (also called CS-186) 1.6mm in diameter placed on the soldered main edge seal to form a gasket, (c) the prepared sample before heating in the oven to 186°C illustrates the upper glass pane (292mm·292mm) placed on the lower glass pane (300mm·300mm) separated by the edge seal and an array of support pillars, and (d) the edge seal made of CS-186 composite and steel reinforced epoxy around the periphery of the sample.

8) The sample was then placed on the hot plate and heated to variable temperatures for improving
evacuation of the cavity in the sample using the vacuum cup connected to the high-vacuum pump-out
system.

9) During evacuation, after 6 hours, the pump-out hole was sealed by heating the CS-186 composite coatedglass square using the cartridge heater fixed inside the vacuum cup as illustrated in the Fig. 10.



Fig. 10. Experimental setup developed for the evacuation of VIG using a vacuum cup connected to avacuum system and the pump-out hole sealed with a glass square.

# 5. Influences of surface temperatures induction on evacuation and pump-out hole sealing of VIG

22

19

23 An ability of VIG to withstand the mechanical stresses is contingent to the strength of the panes of glass 24 and the edge seal. These are the characteristic attributes inherent to the consistent formation of the whole 25 sample, when the cavity vacuum pressure of less than 0.1 Pa and the pump-out hole seal are achieved. Due 26 to the mutual external and internal forces of such a complex procedure, keeping the concentration of the 27 stresses around the pump-out hole area and keeping the minimum possible deflections of the glass surfaces 28 are significant factors in achieving the successful VIG unit. It is important to mention here that uniform 29 temperature distribution and cooling at a slower rate [38] must be introduced because of the thermal 30 expansion mismatch between the glass pane and the main edge seal, i.e.  $8 \cdot 10^{-6}$  C and  $23.5 \cdot 10^{-6}$  C 31 respectively. In this paper, the novel contribution is not only to fabricate VIG but to achieve the prominent 32 vacuum pressure and sustainable surface temperatures with the minimum possible additional stresses. The 33 hot-plate surface temperature and approximate cavity pressure measurements varying with time were 34 performed simultaneously on the seven samples fabricated, each was 300mm 300mm 4mm in size made of 35 K glass. These samples were sealed ,around the periphery of the two glass panes, with the main edge seal 36 10 mm wide made of Sn<sub>56</sub>Pb<sub>39</sub>Zn<sub>3</sub>Sb<sub>1</sub>- AlTiSiCu<sub>1</sub> wt% composite, and a support edge seal, 4 mm wide, 37 made of steel reinforced epoxy. The hot-plate surface temperatures, reported here, were as measured for 38 each sample and for each measurement the temperature controller was set to the appropriate value to study 39 experimentally the influence of hot-plate surface temperatures on the evacuation of the cavity pressure and 40 the pump-out hole sealing of the VIG for the purpose of achieving the viable high-vacuum pressure with 41 the minimum possible stresses. Such stresses are because of shear forces occurred on the edge seal area 42 forcing the glass into curve relative to the centre-of-pane surface. Both glass panes deflect, under the

- 1 induction of temperature differentials, in the same direction that are usually caused during the evacuation
- 2 and pump-out hole sealing process.



Fig. 11. Experimental measurements of hot-plate surface temperature induction and vacuum pressure regimes in
which: (a) Sample A at the set-point of 21°C achieved 0.1 Pa; (b) Sample B at the set point of 40°C achieved 0.05 Pa;
(c) Sample C at the set point of 60°C achieved 0.04 Pa; (d) Sample D at the set point of 80°C achieved 0.03 Pa; (e)

- 6 Sample E at the set point of 95°C achieved 0.02 Pa; and (f) Sample F at the set point of 110°C achieved 0.009 Pa.
- 7

8 Fig. 11a shows the experimental measurements of the approximate cavity pressure under the ambient

- 9 temperature of 21°C. As can be seen, the vacuum pressure of approximately 0.1 Pa was achieved during the
- 10 evacuation. The glass square was heated, using the heating element inside the vacuum cup, gradually to the
- 11 melting temperature of this composite, i.e. 186°C, during evacuation. Due to the temperature gradients on
- 12 the glass panes, the sample-A has experienced increasing level of internal compressive and external tensile
- 13 stresses. This results a small crack on the upper glass around the pump-out hole sealing area occurred after
- 14 10 min during evacuation. It was noticed that that the sample-A must be subjected to an appropriate surface
- 15 temperatures by making sure the surface temperature must not degrade the edge seal. These experimental
- 16 results are in good agreement with the detailed mathematical model and calculations of the predicted
- temperature induced stresses reported by Collins et al. (1992) [24], Fischer-Cripps et al. (1995) [14],
- 18 Lenzen and Collins (1997) [39], Wang et al. (2007) [21] and Wullschleger et al. (2009) [40].

1

2 The Sample-B was fabricated, as shown in Fig. 11b, showing the experimental measurements of the

- 3 approximate cavity pressure of 0.05 Pa at the set-point hot-plate surface temperature of 40°C. The pump-
- 4 out hole of the Sample-B was sealed with glass square but it experienced a small leak, after 15 min of
- 5 evacuation, on the pump-out hole seal due to the insufficient temperature distribution. However, it can also
- 6 be seen that the vacuum pressure was improved as a result of increasing the Sample-B hot-plate surface
- 7 temperature but a proper temperature gradient match was needed between the top-glass surface and the
- 8 heating block inside the vacuum cup. Subsequent stresses observed are tensile on top glass pane and higher
- 9 compressive on bottom glass pane as predicted by Wang et al. (2007) [21].

be reduced by increasing the surface temperatures.

- 10
- 11 Fig. 11c shows the Sample-C temperature/pressure profiles in which the experimental measurements of the
- 12 improved approximate cavity pressure of 0.04 Pa at the set-point hot-plate surface temperature of 60°C
- 13 were recorded. The Pump-out hole of the Sample-C was successfully sealed with glass square after 6 hours
- 14 of evacuation. This is because the layers of adsorbed gaseous molecules as a thin film on the internal
- 15 surfaces within the tubes and vacuum glazing require longer evacuation. The evacuation process time can
- 16 17

18 To further improve the approximate cavity pressure, Sample-D was fabricated in which the approximate

- 19 cavity pressure of 0.03 Pa at the set-point hot-plate surface temperature of 80°C were recorded, as shown in
- Fig. 11d. During the pump-out hole sealing process, it was observed that the Sample-D experienced tensile
- 21 stresses on the top pane whilst compressive on the bottom pane causing glass bending and fractured the
- sample from its edges after 1.5 hours of evacuation during the formation of the pump-out hole seal.
- 23

32

sample nom its edges after 1.5 nouis of evacuation during the formation of the pump-out note sear.

- 24 Such initial experimental investigations show when the hot-plate surface temperature was set to 21°C then 25 it caused difficulty in the formation of pump-out hole seal leading to the growth of crack on the top glass 26 pane. An increase of hot-plate surface temperature facilitates the sealing of the pump-out hole, whilst 27 achieving improved vacuum pressure, but increases the stresses causing bending of the glass panes and 28 produces a risk of fracture to the edge seal. Although the uniform glass surface temperatures are practically 29 not possible due to the limitations of the edge seal temperature for the formation and the mechanical 30 sensitivity of the main edge seal despite the fact the coefficient of thermal expansion of the glass and the 31 edge seal are within their acceptable margins.
- Fig. 11e shows the temperature/pressure measurements of Sample-E in which the approximate cavity
  pressure of 0.02 Pa, at the set-point hot-plate surface temperature of 95°C, was achieved. The pump-out
  hole of the sample-E was successfully sealed but it experienced higher level of internal compressive and
  external tensile stresses, after 20 min of evacuation, caused the fracture of the edge seal.
- 37
  38 To comprehend the limitation of the VIG sample surface temperatures and its maximum achievable
  39 vacuum pressure, Sample-F was fabricated in which the approximate cavity pressure of 0.009 Pa at the set
- 40 point temperature of 110°C were recorded, as shown in Fig. 11f, but cracks occurred, after 4 min of
- 41 evacuation, on to the edge and the pump-out hole areas. It is because of the glass bends due to thermal
- 42 stresses and higher temperature differentials fractured the glass. However, the vacuum pressure was
- 43 improved before 4 min of evacuation by increasing the sample temperature but it also increased the
- 44 stresses, glass deflections, and caused difficulties in sealing the pump-out hole.
- 45

1 A relatively acceptable, based on the aforementioned experimental observations, an improved VIG sample 2 was made at the hot-plate set-point temperature of 50°C, as shown in the Fig. 12. The approximate cavity pressure of 0.042 Pa was achieved and the hot-plate surface temperature and pressure regimes were 3 4 recorded as shown in Fig. 13. It was fabricated after a series of six experiments. In which the influence of 5 hot-plate surface temperatures on the cavity vacuum pressures and their limitations were experimentally 6 studied. A practicable fabrication process was achieved from these experiments and effective sample 7 successively constructed. More than five similar samples of this process having different sizes, re-8 evacuated at the hot-plate set-point temperature of 50°C, were fabricated which validates the recurrent 9 sealing of the pump-out hole and achievable vacuum pressure. The experimental observations show 10 repetitive behaviour of stress patterns across the support pillars indicated a vacuum-tight edge seal, as 11 shown in Fig. 12. A new contribution to this study is that the temperature induces not only stress but it also 12 improves vacuum pressure and achieving the match of hot-plate surface temperature of 50°C for this type 13 of edge seal was a prominent challenge in this study and contribution to the VIG sample. However, the preceding studies have already reported the mathematical modelling of the stresses in vacuum glazing and 14 15 in this paper the repetition was avoided but mainly to follow and validate those predictions experimentally 16 by achieving the successful VIG sample [21, 14]. A careful consideration need to be made when 17 reproducing the VIG construction for larger size and the use of tempered glass could be used and to evaluate the applicability of the obtained results to samples of larger size, current findings are limited to 18 19 smaller size VIG such as the dimensions of 300mm 300mm 4mm or 500mm 500mm 4mm. 20 K-Glass Panes 21 Support Pillar array 22 Main Vacuum edge seal 23 (CS-186 composite) 24 25 Rear view of the pump-out seal 26 27 28 29 30 31 32 33 Secondary edge seal 34 (Epoxy J-B Weld) Glass square pump-out seal 35 protected with Araldite adhesive 36 (b) (a)







2 Fig. 13. The experimental temperature/pressure regimes of sample shown in Fig 12, in which a vacuum

pressure of 0.042 Pa at 50°C was achieved with the successful pump-out hole seal without any leak to the
edge sealing area.

#### 5

7

9

#### 6 6. Thermal performance analyses of the VIG

#### 8 6.1. Validated finite volume modelling approach

An experimentally and theoretically validated finite volume model (FVM) of Fang et al.(2005) [31]; Fang 10 et al.(2006) [15] and Fang et al. (2009) was utilised for the thermal performance analyses of VIG, size of 11 12 300mm 300mm 4mm rebated by 10 mm in a solid wood frame and 10 mm main edge seal. The details of 13 the analytical model are reported in Fang et al. (2006) [22]. A validated set of equations, including the 14 direct depiction of the support pillars incorporated to the FVM, were solved for the fabricated design of 15 VIG at the cavity vacuum pressure of 0.042 Pa. The reason to model only one quarter of the VIG is the 16 symmetrical geometry of the whole sample of VIG under the ISO ambient conditions [41] representing the 17 complete thermal performance. As per ISO (2000) [41] standard, the average air temperatures of the cold 18 and warm sides of the glass panes are set to be 20°C and 0°C, respectively. The inside and outside surface heat transfer coefficients are 7.7 Wm<sup>-2</sup>K<sup>-1</sup> and 25 Wm<sup>-2</sup>K<sup>-1</sup>, respectively. The cylindrical nature of support 19 pillars in FVM is represented as a cube, with square base, support pillar (length of  $\sqrt{\pi a}$ ) having equivalent 20 21 area utilised that conduct the same amount of heat transfer which is a validated approach of Fang et al. 22 (2009) [22]. A higher density of nodes were utilised in the mesh that represents each support pillar to allow 23 maximum possible levels of accuracy in the calculation of heat transfer and again the accuracy is validated 24 in Fang et al. (2005) and the approach is comparable with the results of Wilson et al. (1998) [16] and 25 Collins and Robinsons (1991) [19]. Initial tests of this FVM were performed with the 50.50 nodes 26 distributed on the y and z directions on the glazing surface and with 20 nodes on the x direction. The 27 thermal transmittance at the centre-of-pane for the indium based vacuum glazing with emittance of 0.03 was determined to be  $0.36 \text{ W m}^{-2}\text{K}^{-1}$  with a glass pane thickness of 6 mm. It was found identical with the 28 29 findings of Griffiths et al. (1998) [13] thus this modelling approach is suitable to simulate a practical heat 30 flow with high accuracy of predicting the thermal transmittance of VIG based on the achievable cavity 31 vacuum pressure of 0.042 Pa. The boundary conditions implemented in the finite-volume model of the VIG 32 are listed in Table 2.

1 Table 2

- 2 Boundary conditions implemented in the validated finite-volume model of the VIG.
- 3

Constructional element	Property	Value and/or material type	
Main edge seal	Material	CS186 composite	
	Width	$10 \text{ mm}^{\text{¥}}$	
	Thermal conductivity	$46.49 \text{ Wm}^{-1}\text{K}^{-1}$ *	
Glass pane (Pilkington K type)	Thermal conductivity	$1 \text{ Wm}^{-1}\text{K}^{-1}$	
Emittance	Three surfaces (Hard coating)	0.15/tin-oxide <sup>¥</sup>	
Frame (wood)	Thermal conductivity	$0.138 \text{ Wm}^{-1}\text{K}^{-1}$	
Support pillar	Material	Stainless steel 304	
	Diameter	0.3 mm	
	Height	0.15 mm	
	Pillar separation	24 mm	
	Thermal conductivity	$16.2 \text{ Wm}^{-1}\text{K}^{-1}$	
*Massured thermal conductivities are reported by Memon (2017) [20]			

\*Measured thermal conductivities are reported by Memon (2017) [30].

<sup>¥</sup> In the analyses the comparison is also presented by varying the emittance and edge seal on the thermal performance of VIG

4

5 6.2. Thermal performance of the VIG

6

The centre-of-pane ( $U_{centre}$ ) and total thermal transmittance ( $U_{total}$ ) values of the VIG predicted to be 0.91 7 Wm<sup>-2</sup>K<sup>-1</sup> and 1.05 Wm<sup>-2</sup>K<sup>-1</sup>, respectively. Isotherms of the cold and warm side of the VIG are presented in 8 9 Fig. 14a. This is compared with [28] predictions based on an indium sealed vacuum glazing sample 10 dimensions of 400mm 400mm 4mm with SnO<sub>2</sub> coatings on the inner surface of two glass sheets with a pillar spacing of 25 mm, the U<sub>centre</sub> and U<sub>total</sub> values were reported to be 1 and 1.19 Wm<sup>-2</sup>K<sup>-1</sup>, respectively. A 11 decrease of U<sub>centre</sub> (0.09 Wm<sup>-2</sup>K<sup>-1</sup>) and U<sub>total</sub> (0.14 Wm<sup>-2</sup>K<sup>-1</sup>) values were predicted due to the use of a 10 12 13 mm rebated frame depth and the 10 mm main edge seal covered inside the frame as shown in Fig. 14b. Although the wider layer of edge seal caused increased edge-effects, which results in higher thermal 14 15 transmittance values of the glazing. The total heat transfer can be reduced by reducing the edge seal width and emissivity of the coatings on the inner surfaces of VIG. For example, a 6mm wide indium edge sealed

16 and emissivity of the coatings on the inner surfaces of VIG. For example, a 6mm wide indium economic vacuum glazing was predicted to have  $U_{total}$  and  $U_{centre}$  values of 0.9 Wm<sup>-2</sup>K<sup>-1</sup> and 0.36 Wm<sup>-2</sup>K<sup>-1</sup>,

18 respectively, using soft low emittance coatings [13].



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1	(a) (b)		
2			
3	Fig. 14. (a) isotherms of the one quarter of the VIG where the thickness along the x axial direction is		
4	enlarged by factor of 2.5 compared to the length in y and z direction showing the temperature distribution		
5	from the vacuum edge seal towards the centre-of-pane glazing area. (a) Schematic diagram of the modelled	1	
6	VIG		
7			
8	6.3. An influence of reducing the width of vacuum edge seal and the emittance of inner surface coatings on	l	
9	the thermal performance of VIG.		
10			
11	Fig. 15 shows, that for the VIG size of 300mm 300mm 4mm with an emittance of 0.15, when the edge sea	1	
12	width decreased from 10 mm to 3 mm then the $U_{centre}$ and $U_{total}$ values also decreased from 0.91 Wm <sup>-2</sup> K <sup>-1</sup>		
13	and 1.05 $\text{Wm}^{-2}\text{K}^{-1}$ to 0.81 $\text{Wm}^{-2}\text{K}^{-1}$ (an improvement of 11.0%) and 0.91 $\text{Wm}^{-2}\text{K}^{-1}$ (an improvement of		
14	13.3%) respectively. For the aforementioned size of VIG with an emittance of 0.03, similar decrement of		
15	the edge seal from 10 mm to 3 mm further improved the $U_{centre}$ and $U_{total}$ values from 0.71 Wm <sup>-2</sup> K <sup>-1</sup> and		
16	$0.84 \text{ Wm}^{-2}\text{K}^{-1}$ to $0.62 \text{ Wm}^{-2}\text{K}^{-1}$ (an improvement of 12.7.0%) and 0.71 Wm $^{-2}\text{K}^{-1}$ (an improvement of		
17	15.5%) respectively. These results indicate that further work on reducing the main edge seal width would		
18	improve the thermal transmittance but experimentally reducing the edge seal width has not been possible a	.S	
19	it compromises the integrity and hermeticity of the edge seal of VIG. However, the low-e coatings, such as	\$	
20	silver thin films or transparent nano-structured thin films, could replace SnO <sub>2</sub> coating on K glass as it		
21	improves the thermal transmittances of VIG.		
22			



Fig. 15. Predicted U-value at the centre of glazing and total glazing areas of the 0.3 m by 0.3 m vacuumglazing with edge seal with of 3 mm, 6 mm and 10 mm.

- 6.4. An influence of increasing the size and reducing the width of vacuum edge seal on the thermal
  performance of VIG.
- 3
- 4 Fig. 16 shows, when the glazing size increased from 300mm·300mm·4mm to 400mm·400mm·4mm with
- 5 10 mm edge seal, the  $U_{centre}$  and  $U_{total}$  values decreased from 0.91 Wm<sup>-2</sup>K<sup>-1</sup> and 1.05 Wm<sup>-2</sup>K<sup>-1</sup> to 0.86 Wm<sup>-2</sup>K<sup>-1</sup>
- $6 \quad {}^{2}K^{-1}$  (an improvement of 4.4%) and 0.96 Wm<sup>-2</sup>K<sup>-1</sup> (an improvement of 8.6%) respectively. For the VIG
- 7 with 3 mm wide edge seal, the  $U_{centre}$  and  $U_{total}$  values decreased from 0.81 Wm<sup>-2</sup>K<sup>-1</sup> and 0.91 Wm<sup>-2</sup>K<sup>-1</sup> to
- 8 0.79  $\text{Wm}^{-2}\text{K}^{-1}$  (an improvement of 2.5%) and 0.86  $\text{Wm}^{-2}\text{K}^{-1}$  (an improvement of 5.5%) respectively. These
- 9 results indicate that larger the glazing size the lower the thermal transmittance values. Whilst with a wider
- 10 edge seal, because of its edge effects, the thermal transmittance values are larger than that of the glazing
- 11 with a narrower edge seal. As same as with other kind of edge seal, a larger sized vacuum glazing will

12 provide better thermal performance compared to the one with small size.

- 13
- 14



15

Fig. 16. Predicted U-value at the centre of glazing and total glazing areas of the 300mm 300mm and
400mm 400mm VIG with edge seal width of 3 mm, 6 mm and 10 mm.

18 19

## 20 7. Conclusions

21

Hermeticity of vacuum edge-seal has been the paramount requirement, specifically, in the evolution of
 smart windows. In this paper, a composite (Sn<sub>56</sub>Pb<sub>39</sub>Zn<sub>3</sub>Sb<sub>1</sub>- AlTiSiCu<sub>1</sub> wt%) edge-sealed vacuum insulated
 glazing successfully developed. The main conclusions are summarised into the following four features:

- 25
- 26 (1) A high-vacuum glazing fabrication system, successfully designed and constructed, achieved  $4.35 \cdot 10^{-5}$
- 27 Pa with a modified vacuum cup; this proved to reduce the risk of dislocation of the heating block and the
- 28 degradation of Viton O rings due to unwavering heating required for sealing the pump-out hole with glass
- square inside the vacuum pump during evacuation.
- 30

(2) The microstructural investigations, using FIB-SEM and X-ray CT, of Sn<sub>56</sub>Pb<sub>39</sub>Zn<sub>3</sub>Sb<sub>1</sub>- AlTiSiCu<sub>1</sub> wt%
composite showed negligible traces of micro voids with trapped air inside, when sealed with k-glass, and
homogeneity, when ultrasonically soldered on the glass surface at the vibration frequency of 25-30 kHz
with the tip set-point at 190°C. It led to the development of new methods for the formation of vacuum
edge-seal.

6

7 (3) Experimental investigations of the seven fabricated VIG samples, each of size 300mm 300mm 4 mm, 8 showed that increasing the hot-plate surface temperatures improved the cavity vacuum pressure whilst 9 expediting the pump-out hole sealing process but also increases temperature induced stresses. Successful 10 pump-out hole sealing process of VIG attained at the hot-plate set-point temperature of 50°C and the 11 approximate cavity pressure of 0.042 Pa. More than five similar samples of this process having different 12 sizes fabricated verifies the recurrent sealing of the pump-out hole and cavity vacuum pressure. The 13 experimental observations show repetitive behaviour of stress patterns across the support pillars indicated a 14 vacuum-tight edge seal, one of the vital issue in VIG is its durability and its ageing but in this paper the 15 hermeticity of the composite edge seal itself was analysed by analysing the evacuation time in achieving 16 and maintaining the cavity vacuum pressure before and after evacuation whilst analysing the surface 17 temperature induction influence on vacuum pressure. The durability of the whole sample of VIG itself is 18 significantly important and is a dynamic issue because, despite of successful constructions of VIG, there is 19 always an uncertainty of the degradation of the cavity vacuum pressure because of some gas molecules 20 may remained in the cavity that react when exposed to sunlight and/or under extreme climate conditions for 21 longer time (e.g. after 10 years) due to the development of CO inside the cavity that degrades the vacuum 22 layer. It is apparent that VIG will be exposed to sunlight and need to be designed to sustain at different

- climate temperatures and for over 20 years in order to avoid degradation of vacuum. For this the future
  work recommendation to tackle this issue is to utilise non-evaporable getters in VIG and perform ageing
  tests.
- 26
- 27 (4) A validated finite volume model, incorporating support pillars, employed and calculated the  $U_{centre}$  and 28  $U_{total}$  values of 0.91 Wm<sup>-2</sup>K<sup>-1</sup> and 1.05 Wm<sup>-2</sup>K<sup>-1</sup> respectively for the fabricated VIG sample (size of
- U<sub>total</sub> values of 0.91 Wm<sup>-2</sup>K<sup>-1</sup> and 1.05 Wm<sup>-2</sup>K<sup>-1</sup> respectively for the fabricated VIG sample (size of
   300mm·300mm·4mm) rebated by 10 mm in a solid wood frame at the cavity vacuum pressure of 0.042 Pa.
- 30 Improvements of 11 % (0.81 Wm<sup>-2</sup>K<sup>-1</sup>) and 13.3% (0.91 Wm<sup>-2</sup>K<sup>-1</sup>) in the U<sub>centre</sub> and U<sub>total</sub> values can be
- 31 achieved by reducing the vacuum edge-seal width from 10 mm to 3 mm at the surface coating emittance of
- 32 0.15. For the same size VIG with an emittance of 0.03, when the width of the edge seal decreased from 10
- 33 mm to 3 mm the  $U_{centre}$  and  $U_{total}$  values were predicted to be from 0.71 Wm<sup>-2</sup>K<sup>-1</sup> and 0.84 Wm<sup>-2</sup>K<sup>-1</sup> to 0.62
- 34  $\text{Wm}^{-2}\text{K}^{-1}$  (an improvement of 12.7.0%) and 0.71  $\text{Wm}^{-2}\text{K}^{-1}$  (an improvement of 15.5%) respectively. This
- 35 result indicates that further work on reducing the main edge seal width would improve the thermal
- 36 transmittance values but experimentally reducing the edge seal width has not been possible as it
- 37 compromises the durability of the edge seal of VIG. However, the low-e coatings, such as silver thin films
- or transparent nano-structured thin films, could replace SnO<sub>2</sub> coating on K glass as it improved the thermal
   transmittances of VIG and is suitable for this type of vacuum edge seal.
- 40

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42

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#### Highlights

- Novel design and construction of vacuum insulated glazing (VIG) were discussed.
- A vacuum edge-seal made of composite  $(Sn_{56}Pb_{39}Zn_3Sb_1 AlTiSiCu_1 wt\%)$  was analyzed.
- Influences of temperatures on evacuation and pump-out sealing of VIG were studied.
- A high-vacuum pressure of 0.042 Pa at 50°C surface temperature was achieved with VIG.
- Thermal performance of VIG with surface-coatings and vacuum-edge seal was analysed.

Ctranks