

1 **Microscopic stress analysis of nanoscratch induced sub-**
2 **surface damage in a single-crystal silicon wafer**

3 Ning Huang¹, Ping Zhou¹ and Saurav Goel^{2,3}

4 ¹Key Laboratory for Precision and Non-traditional Machining Technology of Ministry of Education,
5 Dalian University of Technology, Dalian 116024, China

6 ² School of Engineering, London South Bank University, 103 Borough Road, London SE1 0AA, UK

7 ³ University of Petroleum and Energy Studies, Dehradun, 248007, India

8
9 Corresponding author: pzhou@dlut.edu.cn

10 **Abstract**

11 The existing stress criterion assumes the material to be isotropic and only distinguishes
12 elastic, plastic and crack zones to explain the scratching-induced sub-surface damage
13 (SSD) during the contact loading processes such as nanoindentation, nanoscratching and
14 grinding. However, anisotropic single-crystal materials such as monocrystalline silicon
15 and silicon carbide have more diverse defect characteristics and SSD in these materials
16 cannot be well explained and predicted using the existing criterion. In this study, a
17 thorough microscopic characterisation and complementary stress analysis were
18 performed on a single-crystal silicon wafer during nanoscratching. A novel criterion
19 based on mechanism of dislocation multiplication and propagation was proposed and
20 validated, providing a better understanding of SSD prediction in silicon. Compared to
21 conventional SSD models, this new shear stress-based criterion can accurately predict the
22 position and extent of dislocations in silicon. The dislocations layout for scratching along
23 any direction on the (100) surface of Si were further discussed to offer a comprehensive
24 understanding of the effect of anisotropic structure of single-crystals on the SSD. The
25 improved understanding of inelastic deformation in single-crystal silicon, which was
26 revealed by this new model, will have a significant impact on the nanomanufacturing

27 sector by guiding the contact mode experiments (grinding, indentation, machining)
28 towards efficient machining.

29 **Keywords:** Single-crystal silicon; stress criterion; scratch test; sub-surface damage;
30 crystalline defect

31

32 **Abbreviations**

33 α : Boussinesq stress field

34 β : Cerruti stress field

35 γ : Residual blister stress field

36 E : Elastic modulus

37 H : Hardness

38 φ : Half included angle of the tool

39 F_n : Normal load

40 K_c : Fracture toughness

41 b : Half of inelastic zone width

42 ν : Poisson's ratio

43 σ_y : Yield strength

44 SSD: Sub-surface damage

45 ECM: Expanding cavity model

46 ECCM: Expanding cylindrical cavity model

47

48 **1. Introduction**

49 The demand for precision machining of single-crystal silicon is expected to grow at a
50 compound annual growth rate (CAGR) of 9.53% over the next 5 years (2022-2027) ^[1].

51 This is because silicon is among the most widely used semiconductors in various
52 microelectronic applications such as integrated circuits and Photovoltaic (PV) solar cells.
53 Industrially grown wafers of silicon are often subjected to an ultra-precision grinding
54 operation ^[2] to shape them to the desired sizes.

55 However, a grinding process inevitably induces defects in the machined sub-surface of
56 the wafer often referred to as sub-surface damage (SSD), which leads to poor surface
57 integrity and inferior electro-mechanical performance. SSD can further compromise the
58 fatigue life, optical performance and thermal performance ^[3] of the ground component
59 and therefore these defects need to be eliminated by carrying out polishing as a post-
60 grinding operation. The larger the extent of these defects, the more is the extent of post-
61 grinding polishing. Therefore, an insight into understanding the nature of these defects
62 with a view to reduce their extent during grinding can help achieve sustainability during
63 precision grinding.

64 Hitherto, many attempts have been made using molecular dynamics simulations ^{[4][5]},
65 finite element analysis ^[6] and atomic force microscope (AFM) experiments ^[7] to get
66 clarity on these aspects but the mismatch of length/time scale and lack of instruments to
67 directly monitor the sub-surface defects in real time has resulted in a limited progress on
68 this topic. Consequently, the development of a reliable model to predict SSD during
69 grinding has remained a frontal manufacturing challenge in the field of micro/nano scale
70 processing of brittle materials.

71 Single grit scratching is an ideal representation of a contact loading problem, such as
72 grinding. The damage mechanisms that occur during single-grit scratching can provide
73 fundamentally important and new insights into the sub-surface integrity ^[8]. The
74 deformation mechanisms in single-crystal silicon during scratch tests have been studied

75 extensively in the literature ^{[9][10][11]}. Cross-sectional examination of single-crystal silicon
76 during indentation ^[12] and scratch tests ^[10] shows that SSD can involve several events
77 depending on the stress levels experienced by the location and depending on the material
78 being ground, including phase transformation, stacking faults, slip bands and micro-
79 cracks ^[13].

80 At lower loads, the plastic zone may only consist of a phase transformation zone ^[14]. It
81 is now understood that the occurrence of metastable phase transformation in silicon,
82 leading to its amorphisation, is key to exploit ductility in silicon which results in the
83 commonly known brittle-ductile transition ^[11]. As the load increases, dislocations
84 nucleate at the boundary of the phase transformation zone ^[15] and penetrate into the bulk
85 material along specific crystal directions, forming slip bands. As a result, slip bands can
86 usually be observed at an angle of 54.7° to the scratch surface in TEM samples of
87 scratches on the (001) surface of silicon, when the incident electron beam is kept parallel
88 to the $\langle 110 \rangle$ direction, which is the angle between the {100} plane and the {111} plane
89 ^[14]. Researchers have also reported on the critical threshold for dislocation nucleation in
90 silicon ^[16], and the Schmid factor is often used to evaluate the likelihood of activating slip
91 systems in a particular orientation ^[17]. As the load further increases, dislocation
92 accumulation at the intersections of dislocations, such as L-C locks can result in the
93 formation of median cracks, which release the critical loading energy and define the
94 maximum depth of sub-surface damage. Therefore, accurately predicting the depths of
95 various types of damage is an essential foundation for precision grinding process
96 research.

97 Analytical models have been proposed using fracture mechanics approaches to
98 correlate normal load and crack depth. The expanding cavity model (ECM), developed

99 from scratches and indentations, is a primitive effort for understanding the inelastic
100 response of brittle materials. It was postulated that a hemispherical plastic zone engulfed
101 by the contact area creates a stress field that drives ensuing cracks ^[18]. Lambropoulos *et*
102 *al.* ^[19] established the relationship between median crack depth and normal force based
103 on fracture mechanics. Jing *et al.* ^[20] modified the expanding cylindrical cavity model
104 (ECCM) for predicting the size of the plastic zone beneath a single abrasive scratch and
105 the depth of lateral cracks by introducing the Blister stress field. On the basis of
106 Lambropoulos's model and Jing's model, several researches ^{[21][22][23]} have reported an
107 SSD prediction model for the grinding process wherein attempts were made to (i) derive
108 normal load as a function of grit penetration depth using classical indentation hardness
109 theory, (ii) substitute the relationship between normal force and indentation depth into
110 the models to obtain SSD depth as a function of cutting depth during single-grit
111 scratching, and (iii) perform grinding kinematics analysis to establish the SSD prediction
112 model.

113 However, cracks are assumed to generate underneath the plastic zone in the classical
114 ECM, while in single-crystal silicon, cracks emanate at the intersections of crystalline
115 defects. Therefore, the applicability of the classical crack depth model in single-crystal
116 silicon remains to be investigated. Additionally, current models on single-grit damage are
117 typically based on quasistatic loading involving low velocities, around 1 $\mu\text{m/s}$.
118 Considering the strong impact of strain rate on dynamic hardness and fracture toughness,
119 it remains to be determined whether these models are valid for anisotropic materials
120 during high strain-rate deformation such as in grinding.

121 Attempts have been made to consider the effect of material anisotropy by substituting
122 the average Young's modulus and Poisson's ratio from various crystal orientations ^{[24][25]},

123 as the blister stress field equation include material property parameters. However, there
124 is a limitation of the ECCM approach when considering material anisotropy in this model,
125 as the plastic zone is assumed to be semi-cylindrical in ECCM, i.e. the material is
126 isotropic, which clearly differs from the crystallographic orientation-dependent plastic
127 behaviour of anisotropic materials, including the generation and propagation of slip bands
128 into the interior of the material along specific directions ^[12]. This points us to the
129 drawback of the ECCM model and raises a key question whether Jing' model is still
130 applicable to estimate the size of the plastic zone in anisotropic materials like silicon. This
131 question becomes pertinent because to the best of the authors' knowledge, none of the
132 existing analytical models consider anisotropy of inelastic deformation of the material.
133 Besides, there is currently no experimental evidence that provides a direct estimate of the
134 position and extent of dislocation propagation in silicon based on a given crystallographic
135 orientation during contact loading.

136 Based on the literature review, this research paper aims to address the following
137 questions:

- 138 (i) Can the depth of dislocation nucleation be predicted during a scratch tests in an
139 anisotropic brittle material? If so, will the dislocation depth and distribution be the
140 same or different for scratching along different crystallographic orientations under
141 the same load?
- 142 (ii) Can currently available prediction models be reliably used for anisotropic
143 materials such as nanocrystalline silicon? If not, what aspect needs to be further
144 considered to make these models robust?

145 (iii) Can the quasi-static stress field based on the scratch load be used reliably to
146 analyse the extent and nature of defects in a dynamic cutting process such as
147 grinding?

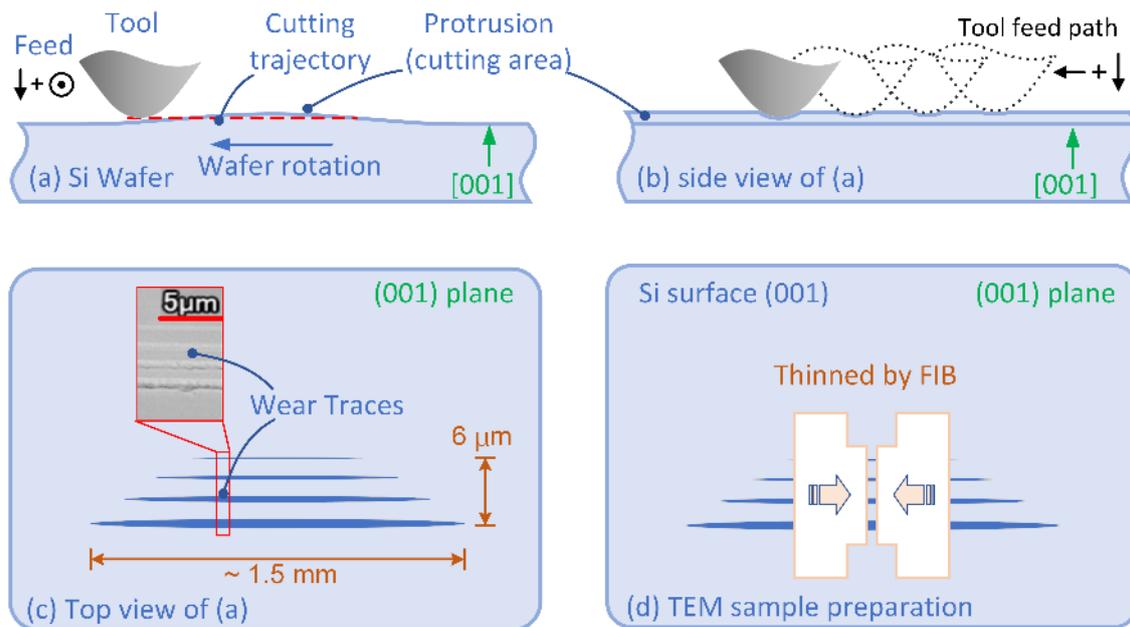
148 To answer these questions, nanoscratching experiments were used in conjunction with
149 engineering analysis and microscopic imaging using advanced microscopic imaging tools
150 such as AFM and TEM to elucidate refreshing new insights into the sub-surface
151 mechanisms in single-crystal silicon. In the discussion section, we have made specific
152 observations on five main points which are (i) directional dependence of amorphisation,
153 (ii) maximum depth of the dislocation during scratching, (iii) maximum depth of the
154 phase-transformed layer, (iv) maximum depth of median crack, and (v) width of the
155 inelastic zone. This paper is expected to provide a guide for gaining a better understanding
156 of the material deformation mechanisms and processing science of single-crystal silicon.

157 **2. Experimental setup and methodology**

158 The scratch tests were conducted on the (001) surface of a commercially available 8-
159 inch p-type monocrystalline silicon wafer using a custom-made scratching apparatus ^[26].
160 The rotary motion of the wafer around its own axis allowed for a scratching speed of up
161 5 m/s, which mimics an actual grinding operation. A protruded area with a curved profile
162 on the wafer shown in [Figure 1\(a\)](#) was chosen for scratching to ensure that scratches were
163 performed with ramped depth-of-cut for measurable scratch length and load. A diamond
164 Berkovich tip with a tip radius of 850 nm was fed radially along the wafer at regular
165 intervals to make parallel wear tracks to avoid duplicate cuts, as illustrated in [Figure](#)
166 [1\(b\)~\(c\)](#). The tip remained static during the contact with the workpiece, so the scratching
167 method used in this paper is essentially passive depth-control. The spacing of the two
168 adjacent scratches was maintained at 2 μm to avoid interference and to ensure that the

169 post-scratching inspection of a single lamella captures all scratches for achieving
 170 consistency in comparison as shown in Figure 1(d).

171 Detailed experimental parameters are shown in Table 1. An atomic force microscope
 172 (XE200, Park systems, Korea) was used to measure the scratched surface using the
 173 tapping mode function. Post-scratching characterization was carried out using
 174 transmission electron microscopy (TEM) (Talos F200X, Fei, America) and focused ion
 175 beam (FIB) (Helios, FEI, America), as per the scheme shown in Figure 1(d). Prior to FIB,
 176 the scratched samples were coated with a polymer film to protect the surface structure
 177 from radiation damage of gallium ion. During TEM, the incident electron beam was kept
 178 parallel to the [110] direction, unless otherwise stated.



179
 180 Figure 1. Schematic illustration of the scratch method (a)~(c) showing wear traces with nanoscale
 181 ramping depth-of-cut and the sample preparation method (d) denoting the area on the scratched region
 182 being chosen for the cross-sectional TEM observation.

183
 184

Table 1 Experimental parameters used for scratching silicon

Sample	Velocity (m/s)	Scratch direction	Number of scratches
A	0.1	(001)[110]	4
B	1	(001)[110]	4
C	1	(001)[010]	4

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3. Results and discussions

188

3.1 Directional dependent amorphisation

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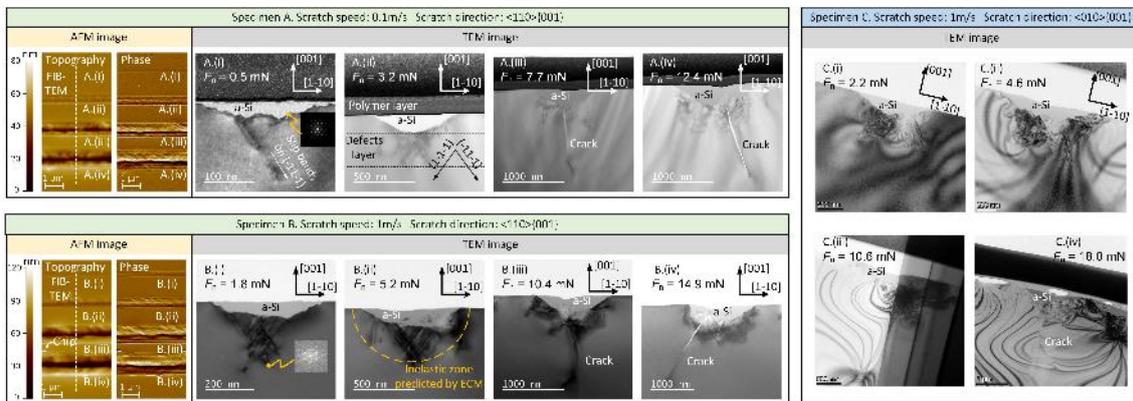
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The AFM and cross-sectional TEM micrographs of residual scratches are depicted in [Figure 2](#). Both topography and TEM images in [Figure 2](#) show that regardless of the scratch speed, the material was squeezed to both sides of the groove, indicating that the scratches were produced by plastic deformation, and that the material removal occurred fully in the ductile-mode [\[27\]](#).



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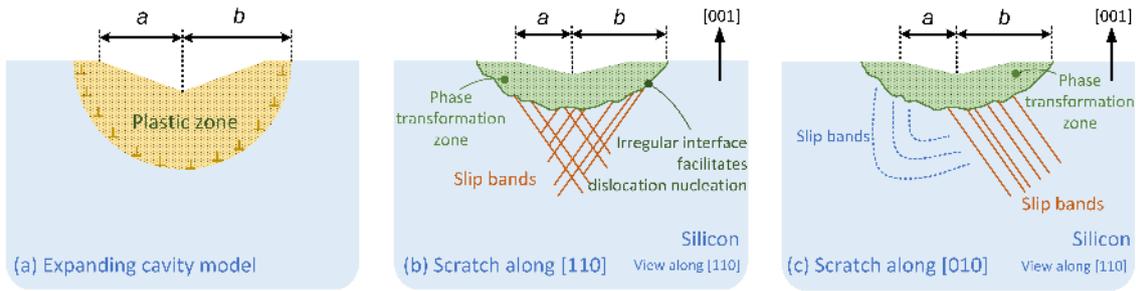
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[Figure 2](#). Scratch topographies inspected by AFM under tapping mode and corresponding detailed TEM observations of sub-surface defects in specimens A, B, and C. The scratches are numbered as A.(i)~A.(iv), B.(i)~B.(iv) and C.(i)~ C.(iv) to clearly show the TEM micrographs at various loads.



200

201 Figure 3. Schematics of scratch groove sub-surface structure in (a) conventional expanding cavity
 202 model, (b) [110] scratch in single-crystal silicon and (c) [010] scratch in single-crystal silicon. Solid
 203 lines in (b) and (c) denote slip bands on {111} slip planes viewed edge-on and dashed and dotted lines
 204 in (c) denote slip bands on {111} slip planes oblique to the [110] projection.

205

206 From the TEM micrographs, the sub-surface deformation of silicon can be seen to have
 207 a directional dependence, as well as a plastic zone and stacking faults generated
 208 underneath amorphous silicon, which are visible as non-equidistant from the axi-
 209 symmetric centre of the indenter in contrast to the semi-cylindrical plastic zone assumed
 210 in the expanding cylindrical cavity model (Hertz theory) highlighted in Figure 3.

211 The TEM results of the scratched specimens revealed two distinct zones, namely, (i)
 212 the phase-transformed region at the upper zone of the sub-surface region, and (ii) the
 213 boundary between the amorphous and single-crystal region, which appears to be an
 214 irregular surface undergoing partial transformation due to differential stress gradients
 215 such that the nucleated dislocations extending into the interior of the material in the {111}
 216 planes remain entrapped after unloading. Tang *et al.* [28] suggested that these defects are
 217 spaced at different intervals as a result of inhomogeneous stress distributions. A few
 218 vertical stacking faults perpendicular to the (100) surface were activated (A(ii)~A(iv),
 219 B(ii)~B(iv)) due to the increased stress levels. The distribution of slip bands in sample C

220 was different from that of sample A and B, as schematically illustrated in Figure 3. This
221 difference is discussed in more detail in the next section on dislocations and slip bands.

222 3.2 Depth of dislocations

223 TEM observations on silicon lamella from Figure 2, as well as from the previously
224 reported indentation [12] and scratch [10] studies on single-crystal silicon samples have
225 shown that the penetration of dislocations into the bulk material within the slip plane is
226 responsible for determining the depth of the plastically deformed zone. According to the
227 well-known Peierls-Nabarro model of dislocations [29], the displacement of a dislocation
228 is closely related to the stress components within the slip plane, because dislocations must
229 overcome an energy barrier to propagate through the lattice. The force required to
230 overcome this resistance is known as *Peierls-Nabarro (P-N)* stress, which varies
231 sinusoidally with respect to the displacement of the dislocation.

232 However, it is acknowledged that the calculation of P-N stress is inaccurate according
233 to Meyers [30], due to the failure of continuum theory at the atomic level (lattice spacing).
234 Therefore, a hypothesis is proposed that the tendency of dislocations to move through a
235 crystal can be described by the stress acting on the dislocation line, and that there exists
236 a critical resolved shear stress, which is the deviatoric component of the applied stress
237 resolved on the slip plane. If the resolved shear stress drops below this critical value, the
238 dislocation cannot propagate. Since the exact value of this critical stress cannot be
239 calculated theoretically, it is suggested that scratching experiments be conducted to
240 determine an experimental-based critical shear stress by evaluating the shear stress at the
241 spatial point where a dislocation terminates. With this critical value, the depth of
242 dislocations induced by mechanical loading can be predicted using theoretically
243 calculated shear stress contours."

244 Peach-Koehler equation ^[30] defines the force F acting on a dislocation line as follows:

$$245 \quad \mathbf{F} = (\boldsymbol{\sigma} \cdot \mathbf{b}) \times \mathbf{l} \quad (1)$$

246 where $\boldsymbol{\sigma}$ is the local stress field, \mathbf{b} is the burgers vector, and \mathbf{l} is the local dislocation line
247 tangent direction. It should be noted that the force \mathbf{F} in Eq. (1) is not a physical force,
248 but rather a means of describing the tendency of a dislocation to move when stresses are
249 present. Eq. (1) shows that the stress components that contribute to the movement of a
250 dislocation depend on factors including the Burgers vectors, the direction of the
251 dislocation lines, and the type of dislocation. Therefore, it is crucial to accurately
252 identify the crystallographic properties of dislocations induced by mechanical stresses
253 in order to determine which stress component is associated with the movement of the
254 dislocation.

255 Danilewsky *et al.* ^[31] and Hänschke *et al.* ^[32] provided a conclusive picture of three-
256 dimensional slip-band arrangements through the use of correlated x-ray diffraction
257 imaging and light microscopy. Their observations showed that the dislocation loops
258 emerged into four $\{111\}$ -planes underneath the indenter during loading (see **Error! R**
259 **eference source not found.**). Each inclined $\{111\}$ slip plane consists of half-hexagonal
260 dislocation loops that multiplies and spread around the scratch, and each dislocation loop
261 consists of two inclined 60° and one screw dislocation with a Burgers vector of
262 $a/2\langle 110 \rangle$ (with lattice constant a). The color code for planes in **Error! Reference**
263 **source not found.** illustrates the $\{111\}$ slip planes around scratch on the (001) surface
264 and the color code for lines in **Error! Reference source not found.(b)** denotes Burgers
265 vector orientations for dislocations with maximum depth. These dislocations were of
266 screw type because the external force generates resolved shear stresses within these slip
267 planes that cause clockwise and anticlockwise rotation on the surface normal ^[33].

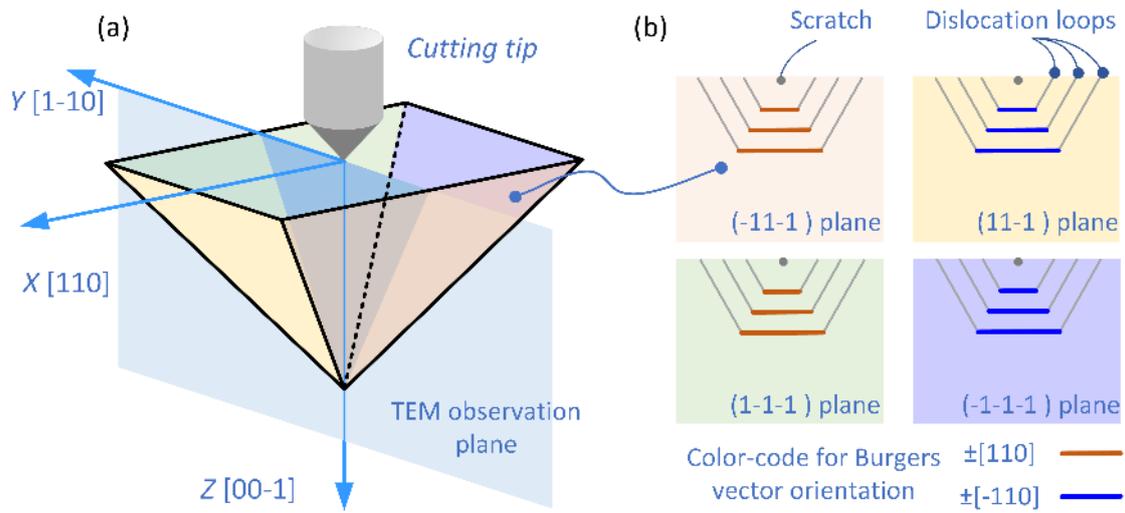
268 According to Eq. (1), shear stress components on the slip plane in the direction of \mathbf{b}
 269 contribute to the gliding force for screw dislocation movement. Therefore, it is natural
 270 to believe that when the shear stress in the $\{111\}$ slip plane along $\langle 110 \rangle$ direction, i.e.
 271 $\tau_{\{111\}\langle 110 \rangle}$, drops to a critical value, the depth corresponding to this shear stress denotes
 272 the maximum depth of the dislocation movement during the scratching of silicon.

273 To verify the above hypothesis, we estimated and obtained the shear stress $\tau_{\{111\}\langle 110 \rangle}$
 274 on the four $\{111\}$ planes (coloured) using analytical calculations. We utilized the work
 275 of Jing *et al.* [20], whose results suggest that the stress field $\boldsymbol{\sigma}$ in the crystalline zone during
 276 single-grit scratch tests can be constructed as a superposition of the Boussinesq stress
 277 field $\boldsymbol{\alpha}$, formed by the normal point force, the Cerruti stress field $\boldsymbol{\beta}$, formed by the
 278 tangential point force, and the sliding blister field $\boldsymbol{\gamma}$, formed by the phase transformed
 279 layer above the crystal defects:

$$280 \quad \sigma_{ij} = \alpha_{ij} + k_1 \beta_{ij} + k_2 \gamma_{ij} \quad (2)$$

$$281 \quad k_2 = f \frac{3\lambda^2}{4\pi^2(1-2\nu)(1+\nu)} \frac{E}{H} \cot \alpha \quad (3)$$

282 where subscripts i and j represent the direction of the stress component, k_1 is the
 283 coefficient of friction, k_2 is the coefficient of the sliding blister field in the phase
 284 transitioned region, and λ is a geometric factor ($\lambda=1$ for an axisymmetric indenter). Here,
 285 a value of the coefficient as $k_1=0.5$ was used [16].



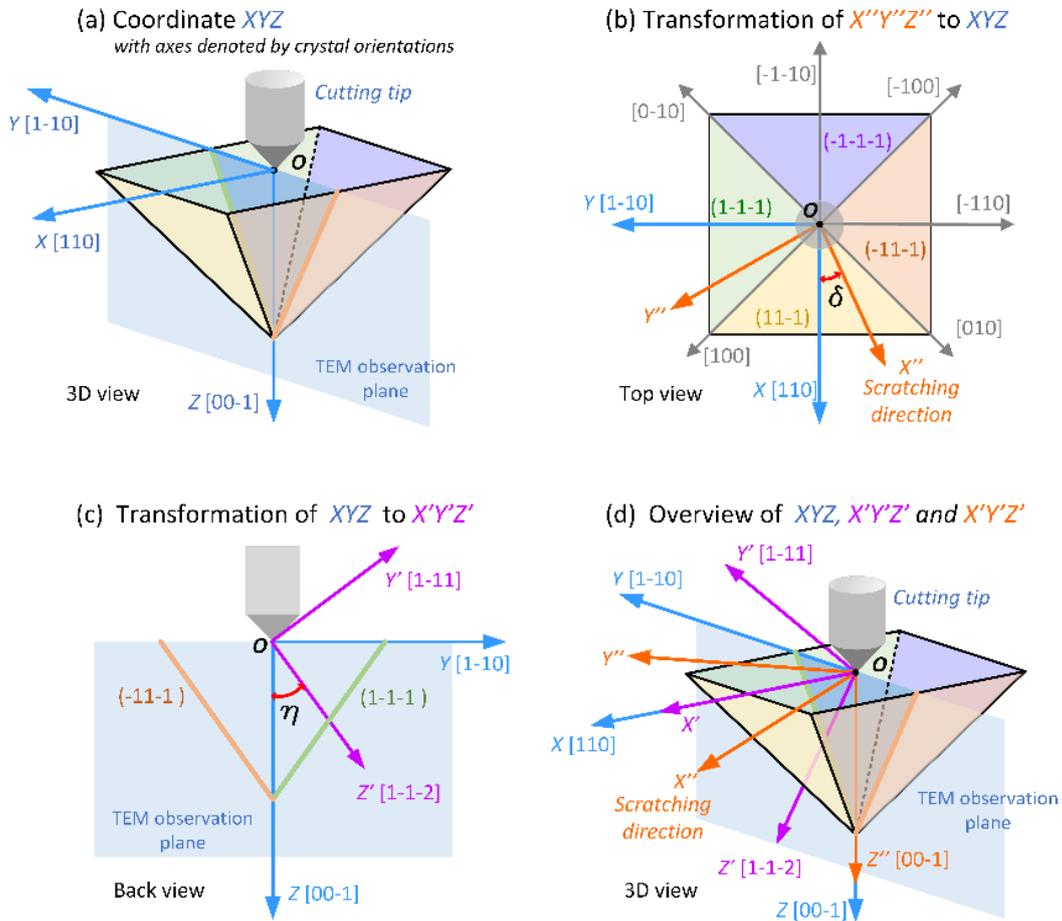
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287 Figure 4. Schematic of (a) formation of slip bands from dislocation loops during scratching and (b)
 288 multiplication and propagation of dislocation loops on the $\{111\}$ slip plane. Each half loop consists of
 289 two inclined 60° and one screw dislocation with a Burgers vector orientation of $\langle 110 \rangle$.

290

291 Two simplifications should be noted regarding Eq. (3). (i) The value of $f \times E/H$ in
 292 Eq. (3) was suggested to be 1.09 for a sharp indenter by Cook and Pharr [34], and as a
 293 result, the coefficient k_2 was considered as 0.1, indicating that the sliding blister field only
 294 accounts for a very small fraction of stress. Additionally, the residual stress of the phase
 295 transformed layer is several orders of magnitude smaller than the stress caused by scratch
 296 load [35]. Therefore, the sliding blister stress field γ caused by the phase transformed zone
 297 has not been considered in this paper. Jing *et al.* [20] derived expression for α_{ij} and β_{ij} in
 298 the cartesian coordinate system. The relevant derived expressions are provided in detail
 299 in Appendix A. (ii) The anisotropy of single crystal silicon, as a crystalline material, is
 300 taken into account in this paper by analysing the slip motion of dislocation loops on slip
 301 planes. However, for the purpose of stress field calculations, we adopted an isotropic
 302 material assumption as stated in Eq. (2) to simplify the analytical calculation of the stress

303 field. A dummy finite element analysis was conducted and it was found that the error
 304 introduced by this simplification is negligible.



305
 306 Figure 5. Schematics of (a) coordinate XYZ denoted by crystal orientation, (b)-(c) coordinate
 307 transformation, and (d) overview of three coordinate systems. The stress field σ is first transformed
 308 from the scratch coordinate system $X''Y''Z''$ to a fixed intermediate coordinate system XYZ by rotating
 309 it by an angle δ around the Z'' axis, which is followed by a rotation around the X axis by angle η ,
 310 resulting in the transformed coordinate system $X'Y'Z'$.

311
 312 The scratch direction OX'' denoted by the orange arrow in Figure 5 can be set to any
 313 direction on the (001) wafer surface to consider material's anisotropy, with the variable δ
 314 in Figure 5 defining the angle between the scratch direction OX'' and the x-axis ([110]
 315 direction). Note that the stress field σ calculated by Eq. (2) is established in the coordinate

316 system $X''Y''Z''$ (hereafter would be referred as scratch coordinate), since OX'' denotes the
 317 scratch direction. To obtain the stress component $\tau_{\{111\}\langle 110\rangle}$, the stress coordinate $X'Y'Z'$
 318 is introduced, with the $X'OZ'$ plane denoting the $\{111\}$ plane and OY' axis denoting the
 319 $\langle 110\rangle$ direction, as shown in Figure 5. The stress field σ is then transformed from the
 320 scratch coordinate $X''Y''Z''$ to the stress coordinate $X'Y'Z'$ using

$$321 \quad \sigma' = \mathbf{T} \cdot \sigma \cdot \mathbf{T}^T \quad (4)$$

322 where \mathbf{T} is the transformation matrix. An intermediate coordinate system XYZ was
 323 introduced to facilitate the calculation of the matrix \mathbf{T} with the following steps: (i) the
 324 scratch coordinate system $X''Y''Z''$ is first rotated around the Z'' axis by an angle δ ,
 325 resulting in the fixed intermediate coordinate system XYZ , as shown in Figure 5(b). (ii)
 326 Then the intermediate coordinate system is rotated around the X axis by an angle η ,
 327 resulting in the stress coordinate system $X'Y'Z'$. The angle η can be determined by
 328 calculating the angle between the Z -axis $[00-1]$ and the Z' -axis $[1-1-2]$, as illustrated in
 329 Figure 5(c). By using this intermediate coordinate system XYZ , Eq. (4) can be rewritten
 330 as:

$$331 \quad \sigma' = \mathbf{T}_2 \cdot \mathbf{T}_1 \cdot \sigma \cdot \mathbf{T}_1^T \cdot \mathbf{T}_2^T \quad (5)$$

332 where \mathbf{T}_1 and \mathbf{T}_2 are transformation matrices for $X''Y''Z''$ -to- XYZ and XYZ -to- $X'Y'Z'$
 333 respectively. The elements of matrix \mathbf{T}_1 was defined in terms of direction cosines of the
 334 angles between scratch coordinate axes OX'' , OY'' , OZ'' and intermediate coordinate axes
 335 OX , OY , OZ :

$$336 \quad \mathbf{T}_1 = \begin{pmatrix} \cos \delta & \sin \delta & 0 \\ \sin \delta & \cos \delta & 0 \\ 0 & 0 & 1 \end{pmatrix} \quad (6)$$

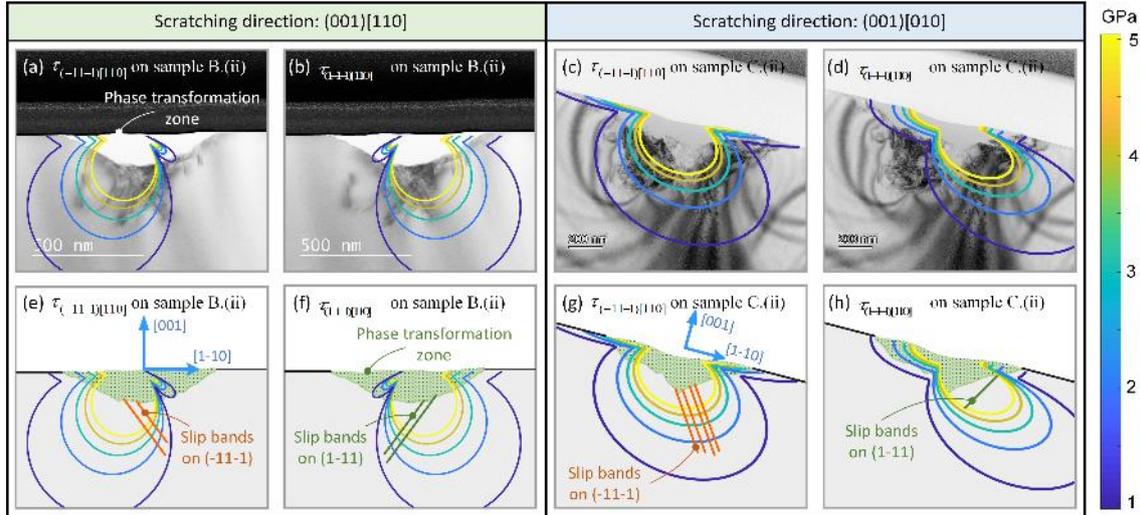
337 Similarly, the elements of matrix \mathbf{T}_2 were defined in terms of direction cosines of the
 338 angles between intermediate coordinate axes OX , OY , OZ and transformed coordinate

339 axes OX' , OY' , OZ' . As the direction of these coordinate axes are defined by the crystalline
 340 orientation, the transformation matrix \mathbf{T}_2 for (-11-1) plane was calculated as:

$$341 \quad \mathbf{T}_2 = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \frac{\sqrt{3}}{3} & -\frac{\sqrt{6}}{3} \\ 0 & \frac{\sqrt{6}}{3} & \frac{\sqrt{3}}{3} \end{pmatrix} \quad (7)$$

342 Using Eq. (2)~(7), the shear stress $\tau_{(-11-1)[110]}$ can be calculated at any point in the space
 343 beneath the scratch tip. Similarly the shear stress $\tau_{(1-1-1)[110]}$ can also be obtained. It is
 344 noteworthy to learn that the behaviour of the slip bands on (-11-1) and (1-1-1) cannot be
 345 predicted from the stress field of the cutting tip at a particular position, as the stress field
 346 within the TEM observation plane (the blue plane in Figure 5) varies continuously with
 347 the cutting tip position during the scratching process. To reveal the relationship between
 348 the shear stress and the slip bands consisting of dislocation loops, we calculated the peak
 349 values of shear stresses $\tau_{(-11-1)[110]}$ and $\tau_{(1-1-1)[110]}$ (hereafter referred as peak stress) in the
 350 TEM observation plane as the tip scratch across this plane, and further compared the stress
 351 contours of peak stress with the TEM images.

352 Figure 6 (a)~(d) present the superimposed stress contours of the peak stress (absolute
 353 value) on top of the TEM images. Two sets of images, samples B(ii) and C(ii), are used
 354 to depict the results for scratches in the direction of [110] and [010] respectively. Each
 355 set contains two identical TEM images, with the left denoting the superimposed $\tau_{(-11-1)[110]}$
 356 and the right denoting superimposed $\tau_{(1-1-1)[110]}$. Fig. 6 (e)~(h) are schematic illustrations
 357 of Fig. 6 (a)~(d) showing a strong correlation between the slip band and the localized
 358 shear stress in Fig. 6 (a)~(d).



359

360 Figure 6. Contours of shear stress (absolute values) $\tau_{(-1-1)[110]}$ and $\tau_{(1-1-1)[110]}$ superimposed on top of
 361 the TEM images (a)(b)(c)(d) and corresponding schematics (e)(f)(g)(h) demonstrating the robust
 362 correlation between the slip band and shear stress.

363

364 The contours in Fig. 6 (a) and Fig. 6 (b) are axisymmetric, as $\tau_{(-1-1)[110]}$ and $\tau_{(1-1-1)[110]}$
 365 are symmetric with respect to the cutting direction of [110]. The same contour lines were
 366 used for all the stress contours in Fig. 6. It is clear from Fig. 6 that slip bands terminates
 367 halfway between the first and second contour lines, despite each sample corresponding to
 368 a different normal load and cutting direction. This demonstrates a strong correlation
 369 between the depth of damage and the magnitude of the shear stress, which qualitatively
 370 verifies of the hypothesis in terms of dislocation distribution. The theoretically resolved
 371 critical shear stress was further benchmarked with the experimental results obtained both
 372 in this paper and previously reported works [14]. These critical stress values are
 373 summarized in Table 2. The data indicates that the critical resolved shear stress $\tau_{\{111\}\langle 110\rangle}$
 374 was within the range of 1.5 ± 0.5 GPa with an approximate minimum magnitude of
 375 1 GPa.

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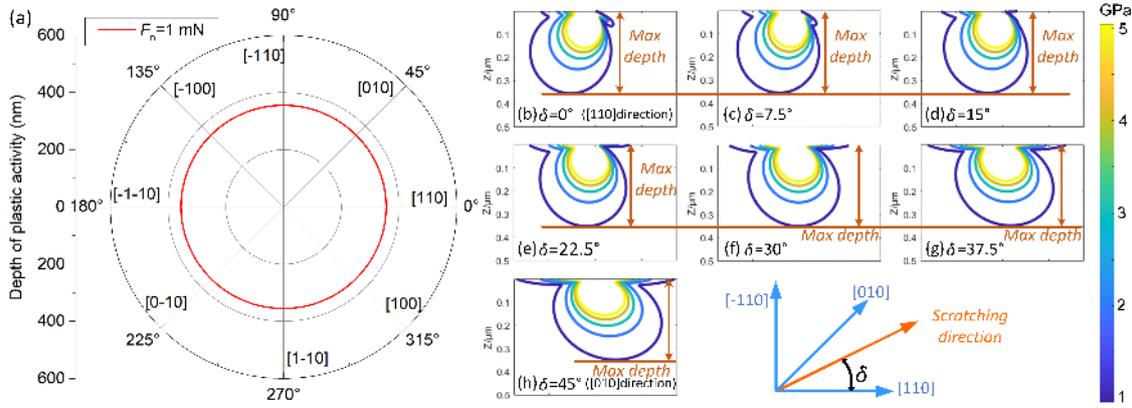
Table 2: Shear stresses at the dislocation termination point in the test samples

Sample	Scratch tip	Direction	Normal load (mN)	Velocity	Minimum shear stress at dislocation position (GPa)
A(i)	Berkovich	(001)[110]	0.5	0.1 m/s	1.4
A(ii)	Berkovich	(001)[110]	3.2	0.1 m/s	1.1
B(i)	Berkovich	(001)[110]	1.8	1 m/s	1.2
B(ii)	Berkovich	(001)[110]	5.2	1 m/s	1.9
C(i)	Berkovich	(001)[010]	2.2	1 m/s	1.6
C(ii)	Berkovich	(001)[010]	4.6	1 m/s	1.7
Huang(i) ^[37]	Conical	(001)[110]	2	0.4 $\mu\text{m/s}$	1.2
Huang(ii) ^[37]	Conical	(001)[110]	4	0.4 $\mu\text{m/s}$	1.3
Huang(iii) ^[37]	Conical	(001)[110]	6	0.4 $\mu\text{m/s}$	1.3

378

379 The same argument explains the variations in the dislocation distribution along
380 different scratching directions as shown in Fig. 3 and Fig. 6. The main observations were
381 (i): In [110] scratches, the slip bands are symmetrically distributed in an 'X' pattern due
382 to axisymmetric shear stresses of $\tau_{(-11-1)[110]}$ and $\tau_{(1-1-1)[110]}$ and slip planes of (-11-1) and
383 (1-1-1). (ii) In [010] scratches, slip bands on (-11-1) planes shifted to the other side of the
384 scratch opposite to that in the [110] scratches, as the stress contour changes. The
385 deformation area at the bottom left of the scratch in the TEM images is due to slip bands
386 on the (11-1) planes, in accordance with the symmetric nature of the four {111} slip
387 planes, which would be easier to understand when viewed in conjunction with Figure 5.
388 (iii) In [010] scratches, the contrast characteristics of the slip bands on the left and right
389 sides of the bottom of the scratch in the TEM image are different because the electron
390 incidence direction [110] in this study is not parallel to the (11-1) planes. These
391 observations provide strong evidence that supports the hypothesis of this investigation,

392 which states that the shear stress acting on a dislocation line in the slip plane determines
 393 the position and extent of dislocations in single-crystal materials.



394
 395 Figure 7. Maximum depth of (a) plastic activity vs. scratch direction and (b) contours of shear stress
 396 $\tau_{(-11-1)[110]}$ indicating maximum dislocation depth as the scratch direction angle δ ranges from 0~45°.

397

398 **Error! Reference source not found.** plots the dislocation depth as a function of scratch d
 399 irection for a given normal load through the stress contours of scratch direction δ in the
 400 range of 0~45°, considering the symmetry of the slip planes illustrated in Figure 5. **Error! R**
 401 **eference source not found.** shows that a change in the direction of scratching leads to a
 402 variation in the shape of the stress contour, which in turn changes the distribution of
 403 dislocations. Specifically, it illustrates that the distribution of shear stress (stress contours)
 404 is direction-dependent, leading to certain cutting directions being "easy" and others
 405 "hard" depending on the crystallographic orientation. Surprisingly, this change does not
 406 alter the theoretical maximum depth of dislocations.

407 Based on the above analyses, an empirical model of the maximum dislocation depth as
 408 a function of the normal load was fitted considering the complexity of the shear stress
 409 analysis:

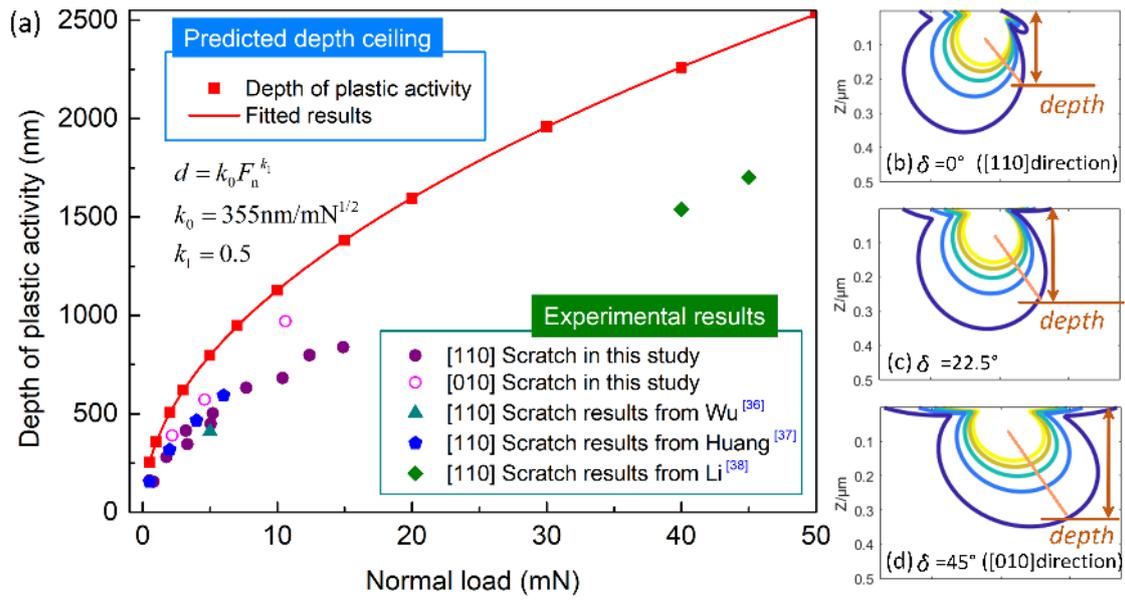
$$410 \quad d = k_0 F_n^{k_1}, k_0 = 355 \text{ nm/mN}^{1/2}, k_1 = 0.5 \quad (8)$$

411 where d is the maximum dislocation depth, F_n is the normal load and coefficient k_0 k_1 are
412 fitted to be $355 \text{ nm/mN}^{1/2}$ and 0.5 respectively.

413 [Figure 8\(a\)](#) establishes the maximum dislocation depth in single-crystal silicon with
414 respect to the scratch normal load. The diagrams together with **Error! Reference source n**
415 **ot found.** provide visual representation of how the distribution and strength of shear stress
416 affects the formation of dislocations at different directions, with comparison to the
417 predicted maximum dislocation depth from the critical shear stress criterion.

418 Two key observations can be made from [Figure 8](#): (i) All experimental results are
419 below the theoretical predictions as expected, especially for the [110] scratches. (ii) The
420 dislocation depth of the [010] scratch is slightly larger than that of the [110] scratch,
421 despite the theoretically predicted maximum depths for these two groups being the same.
422 Both of these phenomena are reasonable because the increase in shear stress τ will result
423 in an exponential increase in the nucleation rate ^[15]. Therefore, the higher stresses within
424 the yellow stress contour promote a greater possibility for nucleation and thus dislocations
425 tends to originate within the area of yellow stress contour where the corresponding depth
426 of critical resolved shear stress is smaller than the predicted maximum dislocation depth,
427 as schematically illustrated in [Figure 8\(b\)~\(c\)](#). The same argument also explains the
428 differences between the [110] and [010] scratches. The depth of critical resolved shear
429 stress corresponding to the area which promotes dislocation nucleation the most in the
430 [110] scratch is smaller than that of the [010] scratch due to differences in the shape of
431 stress contours. This study also observed no significant effects of velocity on the
432 dislocation depth.

433



434

435 Figure 8. (a) Maximum depth of plastic activity vs. normal load obtained from scratch tests in silicon.

436 The red dots and fitted lines denote the predicted maximum dislocation depth according to the

437 proposed critical shear stress criterion. The scatter plots represent experimental data from this study

438 and previously published works [36][37][38]. (b)~(c) are schematic diagrams of shear stress contours at

439 different scratch directions, which explains why experimental results are below the theoretical

440 predictions and why the dislocation depth of the [010] scratch is a bit larger than that of the [110]

441 scratch.

442

3.3 Depth of phase transformed layer

443 As can be seen in Figure 2, the diffraction pattern indicates that the region undergoing

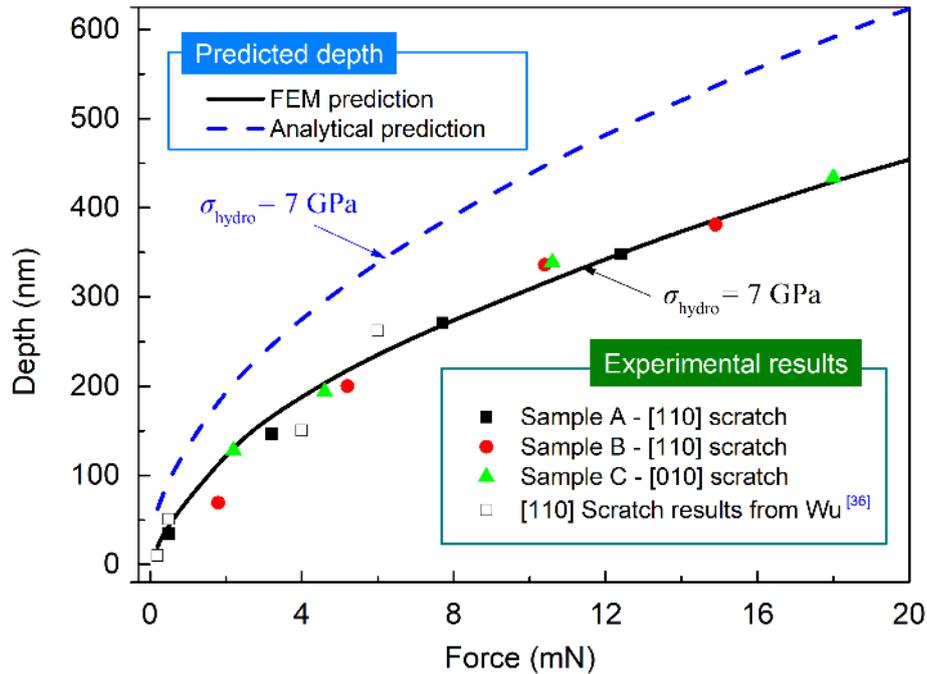
444 phase transformation is composed of amorphous silicon (a-Si) but may also include the

445 formation of growing nanocrystals at deeper penetration depths. The transformed layer

446 forms at the top surface and has an irregular interface between amorphous and crystalline

447 silicon, which is a result of the anisotropy of silicon's atomic lattice structure.

448



449

450 Figure 9. Depth of phase transformed layer as a function of normal load predicted by the critical
 451 hydrostatic stress criterion by analytical and FEM method respectively. Scatter plots denote
 452 experimental data from this study and previously published works [36].

453

454 By analyzing the TEM results of scratched samples, the scatter plot in **Error! R**
 455 **eference source not found.** shows the variation of the depth of the phase transformed
 456 layer with the applied scratch load. According to previous research, the occurrence of
 457 phase transformation is closely linked to hydrostatic pressure, which led us to propose
 458 that the depth of the phase transformed layer can be predicted by calculating the
 459 hydrostatic pressure from the applied load. However, previous studies have reported
 460 different values for the critical hydrostatic pressure at which the high pressure phase
 461 transformation (HPPT) from Si-I to Si-II takes place, ranging from 5.0-8.5 GPa [39], 11.3-
 462 12.5 GPa [40], and 9-16 GPa [37]. To determine the critical stress applicable to scratching,

463 we estimated the hydrostatic stress in the deformation region and plotted the depth
464 corresponding to the critical stress in **Error! Reference source not found.**

465 It is important to note that in this paper, the finite element method (FEM, as discussed
466 in [Appendix B](#)) was used to determine the critical stress value for phase transformation,
467 instead of the analytical method based on [Eq. \(2\)](#). This choice was made for several
468 reasons: (i) using the point loading assumption in the analytical solution would result in
469 errors in the calculation as the depth decreases towards zero, due to the equations used
470 for stress calculations (see [Appendix A](#)); (ii) Furthermore, point loading implies that the
471 contact area is not considered in the analytical calculations, whereas in the calculation of
472 the stress distribution in the phase transformed layer, the contact area may have a strong
473 effect on the calculation results due to the small depth of the phase transformed layer.
474 Therefore, it is considered that the finite element results are more suitable. After
475 specifying the stress threshold, the analytical method based on [Eq. \(2\)](#) was also employed
476 as a comparison to calculate the predicted depth at this threshold stress, leading to the
477 discussion of the following two issues in this section: (i) of the various phase
478 transformation hydrostatic stress thresholds reported previously, which stress value is
479 applicable to the experimental results in this study? (ii) with the critical hydrostatic stress
480 value determined by FEM method, can the analytical method still be used to predict the
481 depth of the phase transformed layer?

482 The high degree of agreement between the FEM-based prediction results and the
483 experimental results in **Error! Reference source not found.** suggests that the depth of t
484 he phase transformed layer can be accurately predicted from the applied load using the
485 hydrostatic stress criterion, with a stress threshold of approximately 7 GPa. However, the
486 analytical method's predictions deviate significantly from both the experimental results

487 and FEM-based predictions under the same stress conditions, which is reasonable given
 488 that the analytical method does not take into account the contact area of the indenter,
 489 unlike the FEM. This highlights the limitations of the analytical method in calculating the
 490 stress field in the near-surface region, due to its simplification of the point load. Therefore,
 491 it is considered unacceptable to use the results of the analytical method to predict the
 492 depth of the phase transformed layer. The experimental results do not indicate any effect
 493 of scratch direction or velocity on the depth of the phase transformed layer.

494 *3.4 Depth of median crack*

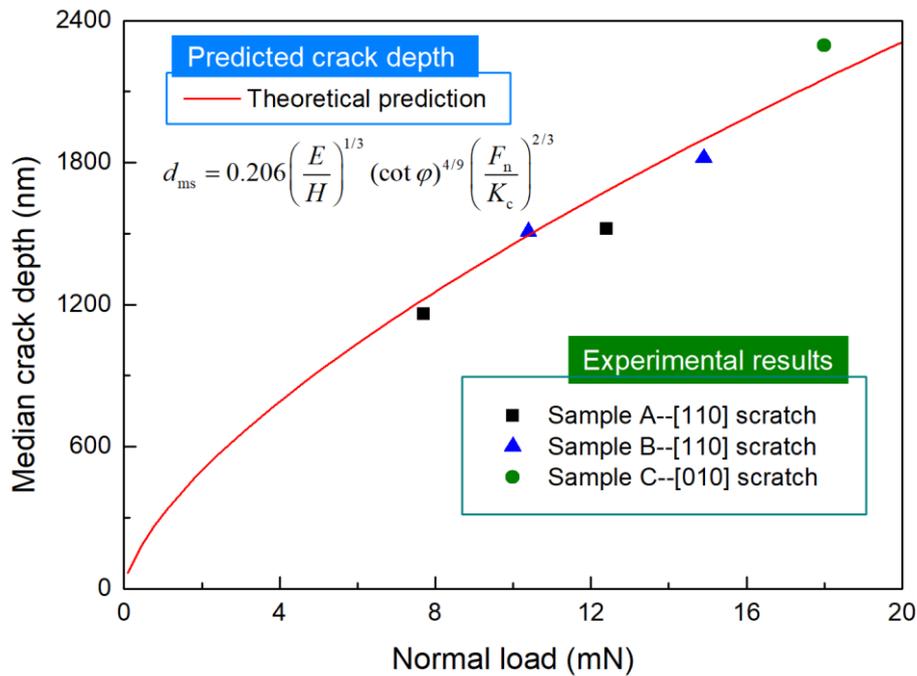
495 As the indentation load increases, microcracks begin to appear at the intersection of
 496 defects. This can be observed clearly in the sub-surface in [Figure 2](#). Many analytical
 497 models that are used to predict crack depth in hard, brittle materials are based on fracture
 498 mechanics theory, such as the widely used Lambropoulos model ^[19]. This model
 499 establishes the relationship between median crack depth (d_{ms}) and normal force as
 500 follows:

$$501 \quad d_{ms} = 0.206 \left(\frac{E}{H} \right)^{1/3} (\cot \varphi)^{4/9} \left(\frac{F_n}{K_c} \right)^{2/3} \quad (9)$$

502 where E , H , F_n and K_c are the Young's modulus, hardness, normal load and fracture
 503 toughness, and φ is half included angle of the tool respectively.

504 For scratch tests performed on silicon (with Young's modulus of $E = 130$ GPa ^[41],
 505 Hardness of $H = 13$ GPa ^[41], Critical angle of $\phi = 72.5^\circ$, Critical stress intensity of K_c
 506 $= 1$ MPa $m^{1/2}$ ^[41]) in this study, [Fig. 10](#) shows the predicted crack depth as a function of
 507 normal load according to the theoretical model, as well as the experimental results for
 508 two different scratch velocities and scratch directions. Similar to the results of the phase
 509 transformation depth analysis, the depth of the median crack is not influenced by the

510 scratch velocity or direction. The comparison of the experimental results with the
 511 theoretical predictions indicates that the model proposed by Lambropoulos ^[19] holds valid
 512 within the scope of this study.



513
 514 Figure 10. Sub-surface crack depth as a function of normal load predicted by the analytical model ^[19].
 515 The symbols in the graph denote the experimental values freshly obtained in this paper.

516
 517 It is worth noting that sample C has the highest normal load corresponding to the first
 518 appearance of median crack among the three groups of samples. This is consistent with
 519 previous qualitative research suggesting that the [100] direction has more plasticity and
 520 ductility than the [110] direction on the (001) cubic face ^[42].

521 Additionally, an interesting phenomenon was observed in the [110] scratches with
 522 median cracks, as an increase in load shifts the angle of the crack from a low Miller index
 523 to a high Miller index, as shown in Figure 2. This phenomenon has also been observed

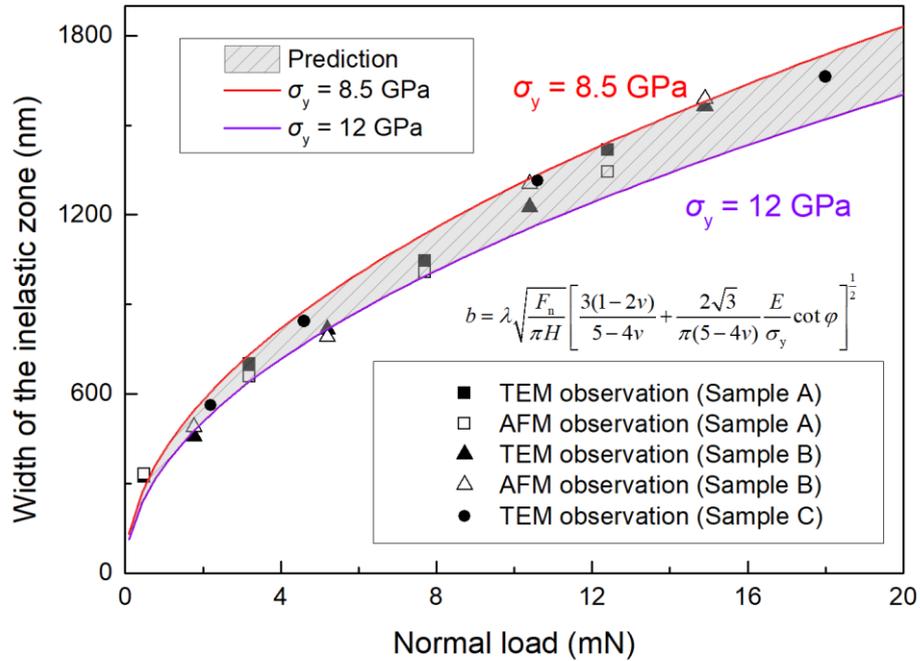
524 during indentation experiments ^[12] on single-crystal silicon and can be explained using
525 the theory of fracture mechanics. The fracture toughness of single-crystal silicon is higher
526 in the direction of high Miller index ^[41]. Thus, a crack in the direction of high Miller index
527 consumes more energy, therefore, at higher loads, a change in the crack angle becomes
528 more energetically favorable.

529 *3.5 Width of the inelastic zone*

530 The width of the inelastic zone of the scratch can be directly measured from cross-
531 sectional TEM micrographs of the scratch (Fig. 6). The results show that the width of the
532 inelastic region is equivalent to that of the phase transformation region under smaller
533 loads. As the load increases and clusters of crystal defects form at the bottom of the phase
534 transformed region, some crystal defects may occasionally extend to both sides of the
535 groove. Due to the random nature of crystal defect propagation, the width of the inelastic
536 zone becomes slightly larger than the phase transformed region. However, given the
537 shallow subsurface damage depth in the ductile-regime grinding and the simplicity of
538 measuring the scratch feature, the width of the phase transformed region on the scratch
539 surface was used as the inelastic region size during the silicon scratching process in this
540 study.

541 Based on this simplification, an AFM was used to measure the dimensions of the
542 inelastic zone of the scratch. Earlier, Figure 2 presented AFM topography and phase
543 micrographs of [110] scratches. Variations in the phase images typically indicate changes
544 in the surface properties of the sample. Since phase transformation of single-crystal
545 silicon alters the mechanical properties of the deformed material (e.g., elastic modulus),
546 it is reasonable to assume that the width of the scratch in the AFM phase image
547 corresponds to the width of the inelastic region on the surface of the scratch. To facilitate

548 comparison between the inelastic zone size measured by the TEM and AFM, the location
 549 of the AFM measurement was aligned with where the FIB-TEM sample was prepared.



550

551 Figure 11. Inelastic zone width as a function of normal load predicted by the analytical model ^[20]. The
 552 scatters represent the experimental results of this paper.

553

554 The measured width of the inelastic zone is shown in Figure 11. The slight deviations
 555 between these two measurement methods can be attributed to the measurement error of
 556 the AFM phase image and the measurement position error of the two methods. Figure 11
 557 shows that the AFM phase images can be used as a substitute for the costly TEM test
 558 when evaluating the width of the inelastic zone for single-crystal silicon scratches.

559 Jing *et al.* ^[20] analytically predicted the size of inelastic regions during scratching using
 560 the ECCM model in indentation, due to its similarity to the scratching process. The
 561 formula is as follows:

$$b = \lambda \sqrt{\frac{F_n}{\pi H}} \left[\frac{3(1-2\nu)}{5-4\nu} + \frac{2\sqrt{3}}{\pi(5-4\nu)} \frac{E}{\sigma_y} \cot \varphi \right]^{\frac{1}{2}} \quad (10)$$

where σ_y is the yield strength, λ is a dimensionless parameter determined by the geometry of the indenter, and ν is the Poisson's ratio. Eq. (10) reveals the relationship between the plasticity zone width and the normal force, tool geometry, and material property parameters. For single-crystal silicon, with a Poisson's ratio ν of 0.28 and yield stress σ_y in the range of 8.5 to 12 GPa [43][44][45], the inelastic zone width as a function of normal load was predicted as the shaded area in Figure 11. The theoretical prediction interval can cover the experimental results well and thus visually confirms the applicability of the Jing's model in the scratching process of single-crystal silicon, that is, the size of the inelastic region is proportional to the square root of the normal load.

The agreement of the experimental results with the theoretical predictions was unexpected because the scratching experiments were performed at scratching velocities of 0.1 m/s and 1 m/s, while the model proposed by Jing *et al.* [20] was developed from the quasistatic indentation tests. It is well-known that the scratch load is positively correlated with the scratch velocity for the same contact area due to the strain rate hardening [46]. The reason why the model is still applicable to single-crystal silicon scratching is that the normal force determines the inelastic zone size of single-crystal silicon during the scratching process for a given tool geometry. Therefore, the effect of scratch velocity on the width is not visible in Figure 11 where the horizontal coordinate denotes the normal load. In addition, the experimental data points from different scratch velocities are clustered on the same curve in Figure 11, indicating that the scratching velocity has little influence on the relationship between the size of the inelastic zone and the normal force, which further justifies the above analysis.

585

586 **4. Conclusions**

587 The existing analytical models available for predicting sub-surface damage depth,
588 critical crack length and the width of inelastic region during its contact loading are based
589 on a significant assumption that the materials are isotropic. A wide variety of engineering
590 materials such as silicon, silicon carbide and diamond used in advanced engineering
591 applications, particularly in the optics and electronics industries are anisotropic and
592 hence, the existing theories cannot be used to readily explain the direction dependent
593 plasticity observed in most sub-surface studies using TEM. This paper proposes a novel
594 shear stress-based criterion to predict the sub-surface damage depth hitherto not reported
595 in the extant literature. The results not only fully demonstrate the feasibility of using the
596 stress field and the appropriate stress criterion to calculate the depth of damage but also
597 revealed the physical mechanisms governing the differences in the deformation at
598 different scratch speeds. In addition, the applicability of inelastic, plastic and median
599 cracking (phase transformation and crack depth) prediction models developed from
600 indentation theory in scratching tests has been also compared. The main findings of the
601 paper were summarised as follows:

- 602 1. Critically resolved shear stress on the slip planes along the direction of Burgers
603 vector can be used to predict both the depth and distribution of dislocations in the
604 sub-surface in silicon as well as the slip bands. In this study, scratch tests were
605 performed on the (001) orientation along the [110] and [010] direction and it was
606 found that the shear stress $\tau_{\{111\}\langle 110\rangle}$ where the dislocation terminates was of the
607 order of 1.5 ± 0.5 GPa. The stress contours congruently overlaid on the TEM images
608 revealed a distinct boundary that limits the travel of a propagated dislocation. In

609 other words, our newly proposed stress based model suggested a pathway for
610 experimental quantification of the post deformation TEM microscopic images.

611 2. The phase transformation depth, crack depth, and the width of inelastic zone, as
612 functions of normal load, were found to be insensitive to changes in the scratch
613 direction, with the exception of dislocation depth and distribution. Only the stress
614 criterion related to crystal orientation needs to take the scratch direction into
615 consideration.

616 3. Due to the close proximity of the phase transformed layer to the contact zone, there
617 is considerable discrepancy in the prediction of the depth of the phase transformed
618 layer by the Boussinesq stress field for a point load, whereas the finite element
619 method, which takes into account the contact area, is able to predict it accurately.

620

621

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635

Appendix A

$$636 \quad \alpha_{xx}(x, y, z) = \frac{P}{2\pi} \left\{ \frac{1-2\nu}{r^2} \left[\left(1 - \frac{z}{\rho}\right) \frac{x^2 - y^2}{r^2} + \frac{zy^2}{\rho^3} \right] - \frac{3zx^2}{\rho^5} \right\} \quad (\text{A-1})$$

$$637 \quad \alpha_{yy}(x, y, z) = \frac{P}{2\pi} \left\{ \frac{1-2\nu}{r^2} \left[\left(1 - \frac{z}{\rho}\right) \frac{y^2 - x^2}{r^2} + \frac{zx^2}{\rho^3} \right] - \frac{3zy^2}{\rho^5} \right\} \quad (\text{A-2})$$

$$638 \quad \alpha_{zz}(x, y, z) = -\frac{3P}{2\pi} \frac{z^3}{\rho^5} \quad (\text{A-3})$$

$$639 \quad \alpha_{xy}(x, y, z) = \frac{P}{2\pi} \left\{ \frac{1-2\nu}{r^2} \left[\left(1 - \frac{z}{\rho}\right) \frac{xy}{r^2} - \frac{xyz}{\rho^3} \right] - \frac{3xyz}{\rho^5} \right\} \quad (\text{A-4})$$

$$640 \quad \alpha_{yz}(x, y, z) = -\frac{3P}{2\pi} \frac{yz^2}{\rho^5} \quad (\text{A-5})$$

$$641 \quad \alpha_{zx}(x, y, z) = -\frac{3P}{2\pi} \frac{xz^2}{\rho^5} \quad (\text{A-6})$$

$$642 \quad \beta_{xx}(x, y, z) = \frac{P}{2\pi} \left\{ (1-2\nu) \left[\frac{x}{\rho^3} - \frac{3x}{\rho(\rho+z)^2} + \frac{x^3}{\rho^3(\rho+z)^2} + \frac{2x^3}{\rho^2(\rho+z)^3} \right] - \frac{3x^3}{\rho^5} \right\} \quad (\text{A-7})$$

$$644 \quad \beta_{yy}(x, y, z) = \frac{P}{2\pi} \left\{ (1-2\nu) \left[\frac{x}{\rho^3} - \frac{x}{\rho(\rho+z)^2} + \frac{xy^2}{\rho^3(\rho+z)^2} + \frac{2xy^2}{\rho^2(\rho+z)^3} \right] - \frac{3xy^2}{\rho^5} \right\} \quad (\text{A-8})$$

$$646 \quad \beta_{zz}(x, y, z) = -\frac{P}{2\pi} \frac{3xz^2}{\rho^5} \quad (\text{A-9})$$

$$647 \quad \beta_{xy}(x, y, z) = \frac{P}{2\pi} \left\{ (1-2\nu) \left[-\frac{y}{\rho(\rho+z)^2} + \frac{x^2y}{\rho^3(\rho+z)^2} + \frac{2x^2y}{\rho^2(\rho+z)^3} \right] - \frac{3x^2y}{\rho^5} \right\} \quad (\text{A-10})$$

648

$$649 \quad \beta_{yz}(x, y, z) = -\frac{P}{2\pi} \frac{3xyz}{\rho^5} \quad (\text{A-11})$$

$$650 \quad \beta_{zx}(x, y, z) = -\frac{P}{2\pi} \frac{3x^2z}{\rho^5} \quad (\text{A-12})$$

651 where $r^2 = x^2 + y^2$ and $\rho^2 = x^2 + y^2 + z^2$, P the normal load and ν Poisson's ratio of the material.

652

653

Appendix B

654 In this study, an indentation finite element model was implemented in the commercial

655 ABAQUS software. Fig. B-1 shows the finite element model for a two-dimensional

656 specimen of $1.0 \times 1.0 \mu\text{m}^2$. The two-dimensional geometry enables an efficient calculation

657 by reducing the number of numerical operations. The indenter is defined as a sphere and

658 is free to move only in the vertical direction. The curvature of the indenter is 850 nm and

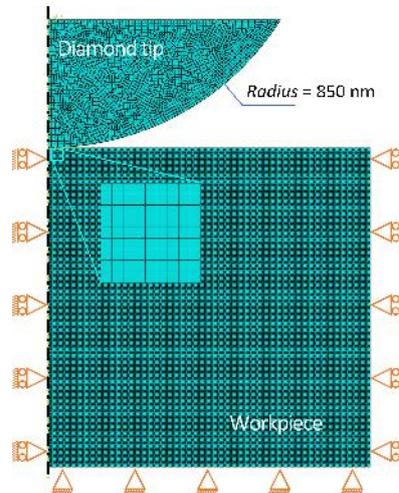
659 is the result of careful measurements, which are reported in our previous paper [26]. The

660 silicon workpiece ($E = 130 \text{ GPa}$ [41], $\nu = 0.28$) was segmented into a 5 nm grid size

661 consisting of 20,000 four-node axisymmetric elements (CAX4R). Rigid walls were used

662 as the boundary of the workpiece as silicon is a hard material and the bottom of the

663 specimen was completely fixed to prevent it from moving.



664

665 Figure B-1. FEM simulation model

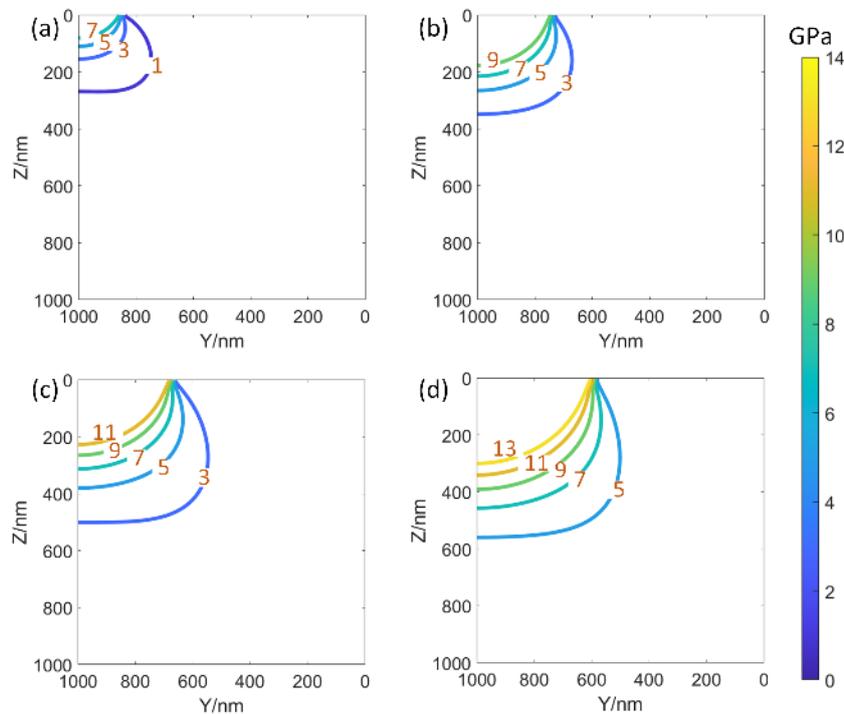
666 Before simulation, the following assumptions are made in the present work:

667 (i) The diamond tip is considered to be a rigid body. The material of the indenter tip is
 668 diamond, which is much stronger and harder than those of silicon. Only slight elastic
 669 deformation occurs in the diamond tip.

670 (ii) Indentation was used to calculate the hydrostatic pressure under normal load during
 671 the scratching process. Analytical calculations based on Eq. (2) show that the tangential
 672 load has a negligible effect on the hydrostatic pressure distribution directly below the
 673 diamond tip.

674 (iii) Only the material elasticity is considered. The TEM results show that the spring-
 675 back of the material under small loads is very significant.

676 The simulation results are demonstrated in Fig. B-2. The stress field is asymmetric, and
 677 the results are shown at the plane of symmetry.



678

679 Figure B-2. Hydrostatic stress (GPa) field in the workpiece at a normal load of (a) 1
 680 mN, (b) 5 mN, (c) 10 mN, and (d) 20 mN, respectively.

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Figure captions

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Figure 1. Schematic illustration of the scratch method (a)~(c) showing wear traces with nanoscale ramping depth-of-cut and the sample preparation method (d) denoting the area on the scratched region being chosen for the cross-sectional TEM observation.

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Figure 2. Scratch topographies inspected by AFM under tapping mode and corresponding detailed TEM observations of subsurface defects in specimens A, B, and C. The scratches are numbered as A.(i)~A.(iv), B.(i)~B.(iv) and C.(i)~ C.(iv) to clearly show the TEM micrographs at various loads.

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Figure 3. Schematics of scratch groove sub-surface structure in (a) conventional expanding cavity model, (b) [110] scratch in single-crystal silicon and (c) [010] scratch in single-crystal silicon. Solid lines in (b) and (c) denote slip bands on {111} slip planes viewed edge-on and dashed and dotted lines in (c) denote slip bands on {111} slip planes oblique to the [110] projection.

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Figure 4. Schematic of (a) formation of slip bands from dislocation loops during scratching and (b) multiplication and propagation of dislocation loops on the {111} slip plane. Each half loop consists of two inclined 60° and one screw dislocation with a Burgers vector orientation of $\langle 110 \rangle$.

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Figure 5. Schematics of (a) coordinate XYZ denoted by crystal orientation, (b)-(c) coordinate transformation, and (d) overview of three coordinate systems. The stress field σ is first transformed from the scratch coordinate system $X''Y''Z''$ to a fixed intermediate coordinate system XYZ by rotating it by an angle δ around the Z'' axis, which is followed by a rotation around the X axis by angle η , resulting in the transformed coordinate system $X'Y'Z'$.

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Figure 6. Contours of shear stress (absolute values) $\tau_{(-11-1)[110]}$ and $\tau_{(1-1-1)[110]}$ superimposed on top of the TEM images (a)(b)(c)(d) and corresponding schematics (e)(f)(g)(h) demonstrating the robust correlation between the slip band and shear stress.

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821 Figure 7. Maximum depth of (a) plastic activity vs. scratch direction and (b) contours of shear stress
822 $\tau_{(-11-1)[110]}$ indicating maximum dislocation depth as the scratch direction angle δ ranges from
823 $0\sim 45^\circ$.

824 Figure 8. (a) Maximum depth of plastic activity vs. normal load obtained from scratch tests in silicon.
825 The red dots and fitted lines denote the predicted maximum dislocation depth according to
826 the proposed critical shear stress criterion. The scatter plots represent experimental data
827 from this study and previously published works ^{[36][37][38]}. (b)~(c) are schematic diagrams of
828 shear stress contours at different scratch directions, which explains why experimental results
829 are below the theoretical predictions and why the dislocation depth of the [010] scratch is a
830 bit larger than that of the [110] scratch.

831 Figure 9. Depth of phase transformed layer as a function of normal load predicted by the critical
832 hydrostatic stress criterion by analytical and FEM method respectively. Scatter plots denote
833 experimental data from this study and previously published works ^[36].

834 Figure 10. Subsurface crack depth as a function of normal load predicted by the analytical model ^[19].
835 The symbols in the graph denote the experimental values freshly obtained in this paper.

836 Figure 11. Inelastic zone width as a function of normal load predicted by the analytical model ^[20]. The
837 scatters represent the experimental results of this paper.

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Tables

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Table 1 Experimental parameters

Sample	Velocity (m/s)	Scratch direction	Number of scratches
A	0.1	(001)[110]	4
B	1	(001)[110]	4
C	1	(001)[010]	4

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Table 2 Shear stresses at the dislocation termination point in the test samples

Sample	Scratch tip	Direction	Normal load (mN)	Velocity	Minimum shear stress at dislocation position (GPa)
A(i)	Berkovich	(001)[110]	0.5	0.1 m/s	1.4
A(ii)	Berkovich	(001)[110]	3.2	0.1 m/s	1.1
B(i)	Berkovich	(001)[110]	1.8	1 m/s	1.2
B(ii)	Berkovich	(001)[110]	5.2	1 m/s	1.9
C(i)	Berkovich	(001)[010]	2.2	1 m/s	1.6
C(ii)	Berkovich	(001)[010]	4.6	1 m/s	1.7
Huang(i) ^[37]	Conical	(001)[110]	2	0.4 $\mu\text{m/s}$	1.2
Huang(ii) ^[37]	Conical	(001)[110]	4	0.4 $\mu\text{m/s}$	1.3
Huang(iii) ^[37]	Conical	(001)[110]	6	0.4 $\mu\text{m/s}$	1.3

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